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TELEDYNE BROWN ENGINEERING HUNTSVILLE AL OPTICAL SYS--ETC F/6 20/13

THERMAL RESPONSE MODEL: MBALL, (U)

SEP 79 W E SMITH

MM-SO-9-79-893

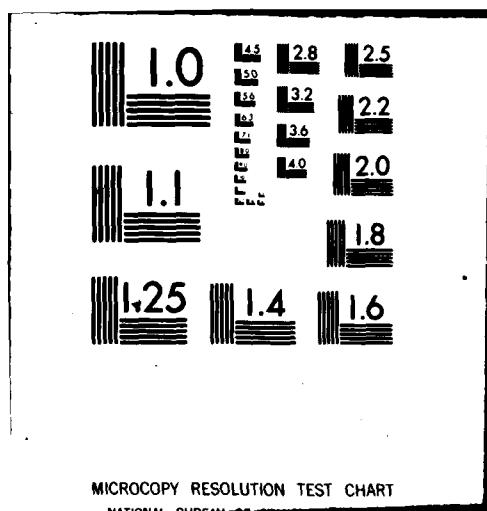
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THERMAL RESPONSE MODEL: MBALL

September 1979

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LETTER OF TRANSMITTAL MSE79-BMDATC-4735



TO: Director  
Ballistic Missile Defense  
Advanced Technology Center  
P. O. Box 1500  
Huntsville, Alabama 35807

Attn: Max Hardwick, ATC-D

FROM: Optical Systems Department  
Systems Division  
Teledyne Brown Engineering  
Cummings Research Park  
Huntsville, Alabama 35807

SUBJECT: Transmittal of MBALL

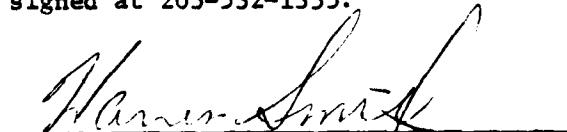
DATE: 7 November 1979

To enable more faithful signature calculations of decoy target concepts applicable to the Optical Discrimination Program, TBE has developed a modification (MBALL) to the Optical Signatures Code (OSC). This modification supercedes all previous versions.

The following is a list of materials to be distributed as specified on the attached page:

- MBALL manual
- OSC Listing (including MBALL in EXOHEAT)
- A short description of the new 6-degree-of-freedom modification to the trajectory program BALLIS
- Update decks of MBALL and the 6-degree-of-freedom option for the OSC VI cycle.

If you have any questions concerning the above, contact the undersigned at 205-532-1355.

  
\_\_\_\_\_  
Warren Smith

LETTER OF TRANSMITTAL MSE79-BMDATC-4735

Enclosures:

Director  
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Advanced Technology Center  
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Letter of Transmittal  
Manual

Attn: Max Hardwick, ATC-D

Ballistic Missile Defense System  
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Letter of Transmittal

Attn: BMDSC-C

Defense Technical Information  
Center  
Cameron Station  
Alexandria, Virginia 22314

Letter of Transmittal  
Manual (2)

McDonnell Douglas Corporation  
5301 Bolsa Avenue  
Huntington Beach, California 92647

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H. Ginell

MIT Lincoln Laboratory  
P. O. Box 73  
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Nichols Research Corporation  
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Suite A  
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Attn: C. Horgen

Sandia Laboratories  
P. O. Box 5800  
Albuquerque, New Mexico 87185

Letter of Transmittal  
Manual  
Listing  
Card Decks

Attn: Bruce Bulmer

## BALLIS EXO 6DOF ANALYTIC SOLUTION

### Insert into OSC VII Basic Manual:

#### BALLIS 6DOF ADDITION: IKIND = 4

In order to save computer time for exoatmospheric 6 degree of freedom trajectories where two moments of inertia are equal, an analytic solution to Euler's equations with no external torques has been developed. Such a solution is valid for axi-symmetric bodies above the top of BALLIS's atmosphere ( $\sim 10^6$  feet). The rationale for this modification is to reduce the computer time required by BALLIS. The same input is required for this option as for option: IKIND = 3, IKIND6 = 0. The option is accessed by inputting IKIND as 4 on Card 4A. Cards 4B, 5 and 6 are also required.

### Insert into Table 4-11 Basic Manual:

IKIND

.....

=4 Exo-axisymmetric 6DOF Analytic Solution  
Cards 5 and 6 required.

14) MN-SD-9-79-893

(6) THERMAL RESPONSE MODEL: MBALL

By

W. E. / Smith

September 1979

Sponsored By

BALLISTIC MISSILE DEFENSE ADVANCED TECHNOLOGY CENTER  
DEPARTMENT OF THE ARMY  
HUNTSVILLE, ALABAMA

Prepared By

OPTICAL SYSTEMS DEPARTMENT  
SYSTEMS DIVISION  
TELEDYNE BROWN ENGINEERING  
HUNTSVILLE, ALABAMA

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## ABSTRACT

MBALL is a subroutine of EXOHEAT for surface temperature calculations for convex objects of light to intermediate thermal mass (in-depth heat conduction is not important). It includes surface conduction and internal radiative coupling between stations, using the natural fluxes (Sun, albedo, earthshine) calculated in EXOHEAT. MBALL must be used with the OSC VI BASIC option.

## APPROVED BY:

  
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Accession For

NTIS GENIUS  
DOD Test  
Uncontrolled  
J. A. Johnson  
*Litter Drift*  
FBI

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## 1. INTRODUCTION

### 1.1 MBALL OVERVIEW

Subroutine MBALL was developed to provide target surface temperatures in an exoatmospheric environment for balloon and replica shapes of light to intermediate thermal mass. These temperatures are used by the Optical Signatures Code (OSC) (Ref. 1) to generate long wavelength infrared (LWIR) signatures for threat discrimination analysis. It is necessary that vehicle geometry, deployment, and external flux information be supplied by the OSC VI to MBALL, which performs temperature calculations for thermally light objects.

The OSC execution sequence is shown in Figure 1-1. MBALL is located in the EXO/ENDO thermal response block as a subroutine of EXOHEAT.

### 1.2 MBALL CAPABILITIES

Table 1-1 is a summary of MBALL's capabilities. For each point of the vehicle surface at which temperatures are calculated, MBALL uses averages of in-depth material properties (assuming a layered skin of different materials) and computes an average temperature as if the skin were made up of one homogeneous material. This temperature is then assigned to the surface of the vehicle for signature generation.

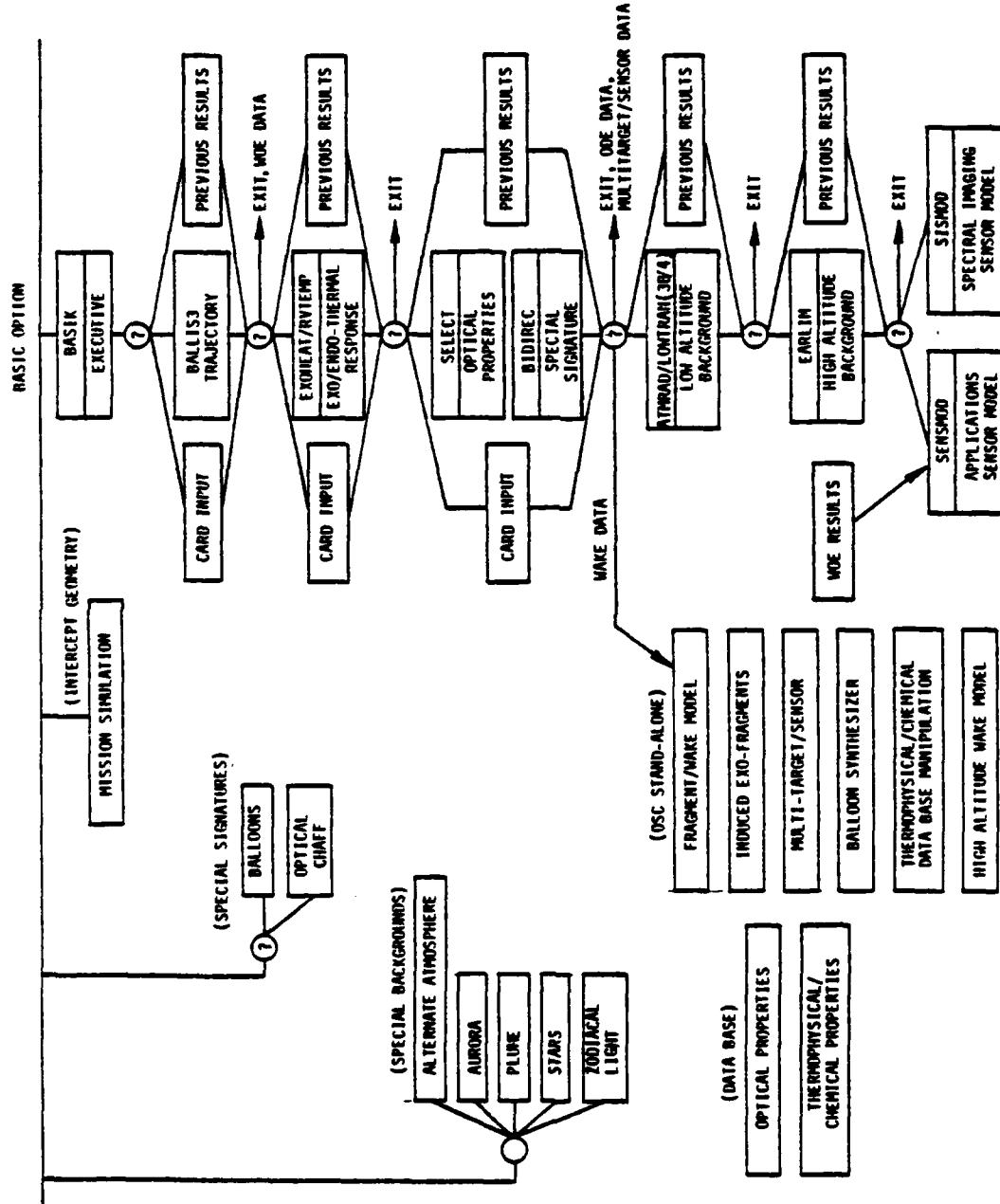


FIGURE 1-1. OSC VI PROGRAM STRUCTURE

TABLE 1-1. MBALL CAPABILITIES

Targets	<ul style="list-style-type: none"> <li>● Thin to Intermediate Thermal Mass</li> <li>● Replica or Balloon Shapes</li> </ul>
Thermal Fluxes	<ul style="list-style-type: none"> <li>● Solar</li> <li>● Albedo</li> <li>● Earthshine</li> <li>● Molecular</li> </ul> <p style="text-align: right;">From EXOHEAT Subroutine (EXOENV)</p>
Station Coupling	<ul style="list-style-type: none"> <li>● Internal Radiation</li> <li>● Thermal Conduction: Longitudinal and Transverse</li> </ul>
Thermal Response	<ul style="list-style-type: none"> <li>● Phase Change Capability</li> <li>● Thermal Properties Updated with Temperature</li> </ul>
Data Base	<ul style="list-style-type: none"> <li>● Earthshine Data: Models Based on NIMBUS Observations (Ref. 2)/LOWTRAN4 Calculations</li> <li>● Thermophysical Properties: OSC</li> <li>● Thermophysical/Chemical Data Base: OSC</li> </ul>
Options	<ul style="list-style-type: none"> <li>● Radiative Equilibrium (Thermal Mass = 0)</li> <li>● Open Surfaces</li> </ul>

## 2. FUNCTIONAL ANALYSIS

Target surface temperatures are determined in the OSC by EXOHEAT for exoatmospheric conditions (greater than 400 kft) or by RVTEMP for endoatmospheric conditions (less than 400 kft). These programs need target position, velocity, and deployment information as a function of trajectory time to compute the temperatures. This information is supplied by BALLIS or by the user. The temperatures are then used by BIDIREC to generate radiometric signatures, with material optical properties supplied by SELECT. BALLIS, EXOHEAT, RVTEMP, SELECT, and BIDIREC are all contained in the OSC and are called by the OSC program BASIC\$, depending on the user input.

MBALL is a subroutine of EXOHEAT and replaces EXOHEAT in-depth temperature calculations (i.e., replaces Subroutine EXOTMP) for objects of light to intermediate thermal mass. Heat flux calculations are performed in EXOHEAT (Subroutine EXOENV) for a given ballistic trajectory and vehicle geometry to determine external heating rates (from Sun, albedo, and Earth emission) and these are passed to MBALL, bypassing EXOHEAT's temperature subprograms. The vehicle geometry required by MBALL is computed in RVSNTH, a subroutine of BASIC\$. Previously, RVSNTH was used only with the endoatmospheric heating routine, RVTEMP, but has been modified to supply MBALL with the necessary parameters. RVSNTH does not calculate SELECT or BIDIREC data when MBALL is called, however, so the inputs to SELECT and BIDIREC must be supplied by the user. The calculations that determine the external heating fluxes are described in the EXOHEAT manual (Ref. 3) (written before the MBALL option existed). Reference 1 contains the input to SELECT and BIDIREC. Figure 2-1 shows the EXOHEAT program flow. A description of how MBALL is called in the BASIC option is in Section 4.

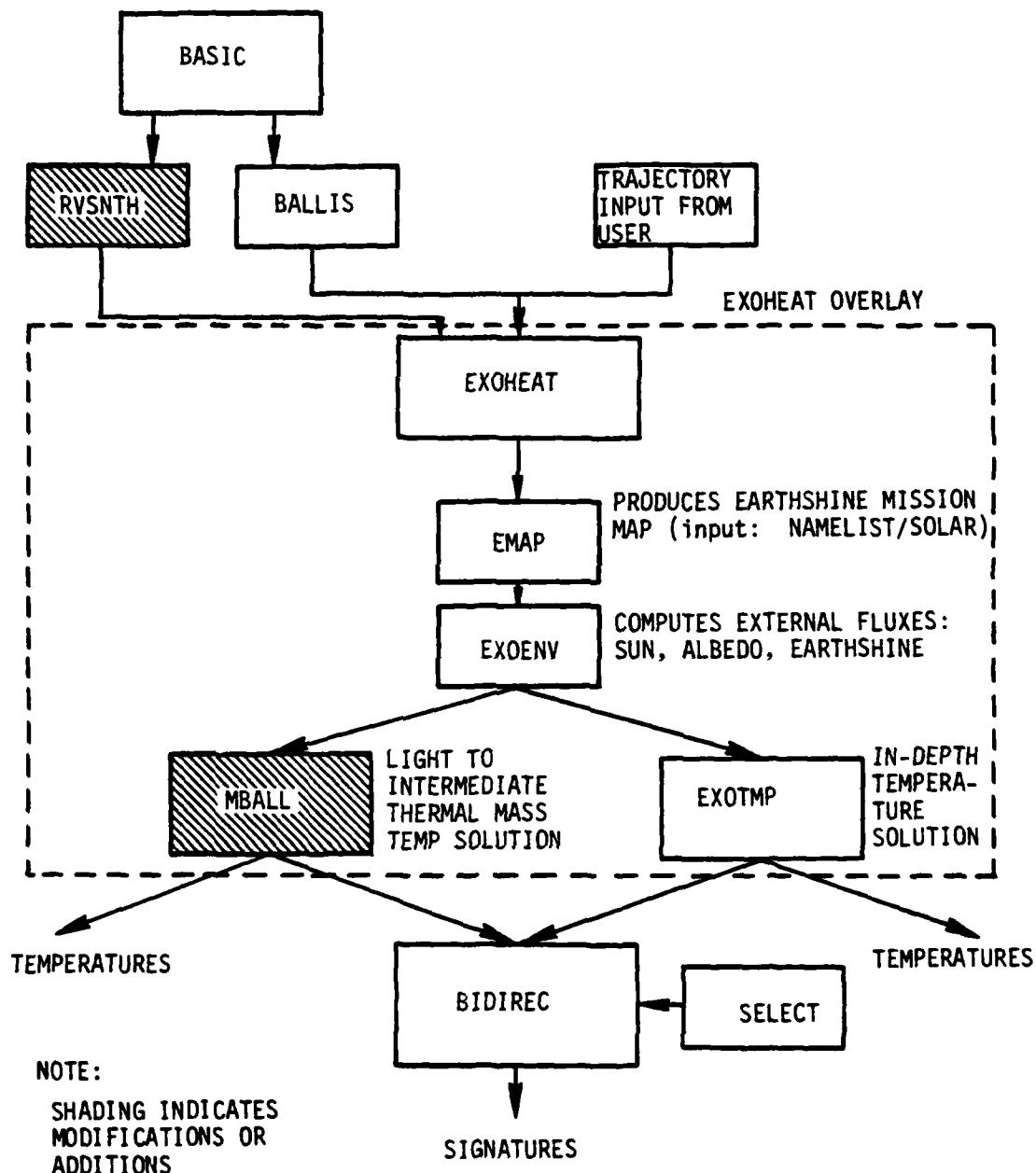


FIGURE 2-1. EXOHEAT PROGRAM STRUCTURE

### 3. THEORY

#### 3.1 HEAT EQUATION

The heat equation expressing the time rate of change of temperature  $T$  at any point in a material of density  $\rho$ , thermal conductivity  $k$ , and specific heat  $c_p$ , with an internal energy source density  $S$ , can be written:

$$\rho c_p \frac{\partial T}{\partial t} = \nabla \cdot (k \vec{\nabla T}) + S \quad (3-1)$$

where the term  $k\vec{\nabla}T$  can be thought of as an energy density, whose divergence gives an energy source density. The term  $\vec{\nabla}T$  is the spatial gradient of the temperature.

In MBALL, the thermally light skin of a replica or balloon shape is modeled by dividing it into  $N$  pieces defined by the user, where each piece has associated with it an outer surface, an inner surface, and up to four surfaces that adjoin adjacent pieces. The outer surface communicates with the external environment via radiative coupling, the inner surface communicates with the other interior surfaces in the same manner, and the adjoining surfaces communicate through heat conduction. Each piece also has an associated density, specific heat, and thermal conductivity. The values chosen for these quantities for a given piece, say the  $j$ th piece, are an appropriate average of the exact values taken over the volume of the piece. By integrating Equation 3-1 over the  $j$ th piece with this point in mind (the subscript  $j$  refers to average values), one obtains:

$$\int_{V_j} \rho_j c_{pj} \frac{\partial T_j}{\partial t} dV_j = \int_{\substack{\text{Areas} \\ \text{Bounding } V_j}} (k \vec{\nabla}T) \cdot \hat{n}_j dA_j \quad (3-2)$$

where  $\hat{n}_j$  is a unit normal pointing out of surface  $dA_j$  of piece  $j$ , and  $S = 0$ ; i.e., the assumption is made that no independent sources of energy (batteries and wires) exist in the skin of our model. Because the

material quantities are averages, they can be taken as constant over the volume of integration. The area integral can be resolved into its component parts:

$$\rho_j c_{pj} V_j \frac{\partial T_j}{\partial t} = \int_{A_j \text{ outer surface}} (\vec{kV}T) \cdot \hat{n}_j dA_j + \int_{A_j \text{ inner surface}} (\vec{kV}T) \cdot \hat{n}_j dA_j \\ + \int_{A_j \text{ adjoining surfaces}} (\vec{kV}T) \cdot \hat{n}_j dA_j . \quad (3-3)$$

The boundary conditions for the outside and inside surfaces of the jth piece can be written as follows, where the jth piece is assumed to radiate as a greybody with temperature  $T_j$  and emissivity  $\epsilon_j$ :

$$\hat{n}_j \cdot \vec{kV}T = -\epsilon_j \sigma T_j^4 + Q_{ext,j} \text{ (outside)} \quad (3-4)$$

$$\hat{n}'_j \cdot \vec{kV}T = -\epsilon'_j \sigma T_j^4 + Q_{int,j} \text{ (inside)} \quad (3-5)$$

where prime marks denote internal normals and emissivities, and  $\sigma$  is the Stefan-Boltzmann constant.  $Q_{ext,j}$  is the average power/area incident on the exterior of piece j and is due to the external environment.  $Q_{int,j}$  is the average power/area striking the interior of piece j and is due to the other radiating pieces.

Substituting Equations 3-4 and 3-5 into Equation 3-3 for the appropriate surface integrals yields:

$$\rho_j c_{pj} V_j \frac{\partial T_j}{\partial t} = \int_{A_j \text{ outer}} (Q_{ext,j} - \epsilon_j \sigma T_j^4) dA_j \\ + \int_{A_j \text{ inner}} (Q_{int,j} - \epsilon'_j \sigma T_j^4) dA_j + \int_{A_j: \text{ adjoining}} (\vec{kV}T) \cdot \hat{n}_j dA_j . \quad (3-6)$$

Because these are averages, and assuming  $A_{outer} = A_{inner}$   
(thin skin):

$$\rho_j c_{pj} V_j \frac{\partial T_j}{\partial t} = Q_{ext,j} A_j + Q_{int,j} A_j - (\epsilon_j + \epsilon'_j) \sigma T_j^4 A_j + \sum_{\substack{\text{adjoining} \\ \text{areas}}} A_{\text{adjoining}} k_{eff} \frac{T_{\text{adjoining}} - T_j}{\Delta X_{j:\text{adjoining}}} \quad (3-7)$$

where the conduction integral of Equation 3-6 has been approximated by a sum over all adjoining surfaces of the piece j. The terms  $A_{\text{adjoining}}$ ,  $k_{eff}$ ,  $T_{\text{adjoining}}$ , and  $\Delta X_{j:\text{adjoining}}$  refer to the contact area between piece j and its neighbor, a properly averaged thermal conductivity between piece j and its neighbor (discussed below), the average temperature of the adjoining piece, and the distance between the centers of j and its neighbor, respectively.

It is convenient to rewrite Equation 3-7 in the following way to simplify the bookkeeping:

$$\rho_j c_{pj} V_j \frac{\partial T_j}{\partial t} = Q_{ext,j} A_j + \sum_i M_{ij} T_i + \sum_i K_{ij} T_i \quad (3-8)$$

where the matrices  $M_{ij}$  and  $K_{ij}$  contain the radiation and conduction terms, respectively, which are discussed in detail below. The sums are over all pieces, including the jth.

Because of the need to model materials that have rapidly changing specific heats as a function of temperature (i.e., water near 0°C), it is necessary to generalize the left-hand side of Equation 3-8:

$$\rho_j c_{pj} V_j \frac{\partial T_j}{\partial t} \longrightarrow \rho_j V_j \frac{\partial}{\partial t} \int_{T_{0j}}^{T_j} c_{pj}(T'_j) dT'_j \quad (3-9)$$

where  $T_{oj}$  is the initial temperature of piece  $j$ , and  $T_j$  is the temperature after some time  $t$ . Thus, the equation solved for  $T_j$  by MBALL can be written:

$$\rho_j V_j \frac{\partial}{\partial T} \int_{T_{oj}}^{T_j} c_{pj} (T_j'') dT_j'' = Q_{ext,j} A_j + \sum_i M_{ij} T_i^4 + \sum_i K_{ij} T_i . \quad (3-10)$$

The linearization of the left- and right-hand sides necessary for an iterative solution for  $T_j$  is described in detail in Section 3.

### 3.1.1 External Sources

There are four natural external sources of radiation considered: sunshine, albedo, earthshine, and molecular collisional heating. These parameters are calculated in EXOHEAT and are passed to MBALL. Reference 3 contains a complete discussion of these quantities.

- Sunshine - A solar exoatmospheric irradiance of 1353 W/m<sup>2</sup> is assumed. The Sun position is input in one of three ways:
  - ▲ Subsolar point input: latitude and longitude
  - ▲ Subsolar point calculated from month, day, GMT
  - ▲ Subsolar point calculated from month, day, local (24-hour) time at a given longitude
- Albedo - The reflectivity ( $\alpha$ ) of the sunlit Earth is assumed to be 0.4. The albedo radiance of a sunlit element of the Earth is

$$N_a = \frac{\alpha S \cos \theta}{\pi} \quad (3-11)$$

where  $S$  is the solar constant and  $\theta$  is the angle between the Earth normal and the direction to the Sun.

- Earthshine - Six Earth radiance maps are available to EXOHEAT, corresponding to day/night conditions for winter, summer, and equinox. Each map corresponds to a 5- by 5-deg latitude, longitude grid. The seasons are determined by the month, and Sun position determines the time of day. Reference 2 contains a magnetic tape appendix containing earthshine data. The earthshine map may be modified with NAMELIST/SOLAR input (Ref. 3, p. 6)
- Molecular - At altitudes below  $10^6$  ft, a molecular collisional heating flux is used of the value

$$Q_{\text{air}} = \frac{1}{2} \rho_{\text{air}} V^3 (\hat{n} \cdot \hat{v}) \quad (3-12)$$

where  $\rho_{\text{air}}$  is the air density and  $V$  is the velocity of the target.

The effect that the above contributions have on the energy incident on a given area element of the target depends on the geometry of the Sun, Earth, and target configuration, and where the area element is located on the target. Only the flux component normal to the area element is considered. Reference 3 contains a detailed description of how this problem is treated. Figure 3-1 shows the geometry.

EXOHEAT has the option of calculating three different flux averaging methods (Table 3-1):

- Instantaneous - Energy incident on a given target element is that which is determined by the instantaneous geometry (i.e., position of the Sun and Earth, and location of the element on the target)
- Roll-Averaged - Energy incident on a given target element is the normalized average of the energy the element would see through a full revolution about the target longitudinal axis; therefore, all elements lying in a ring centered on the body axis experience the same incident energy.
- $4\pi$  Average - Incident energy is the normalized average of the energy that the element would see through a "tumble" over  $4\pi$  steradians. All elements on the target receive the same incident energy.

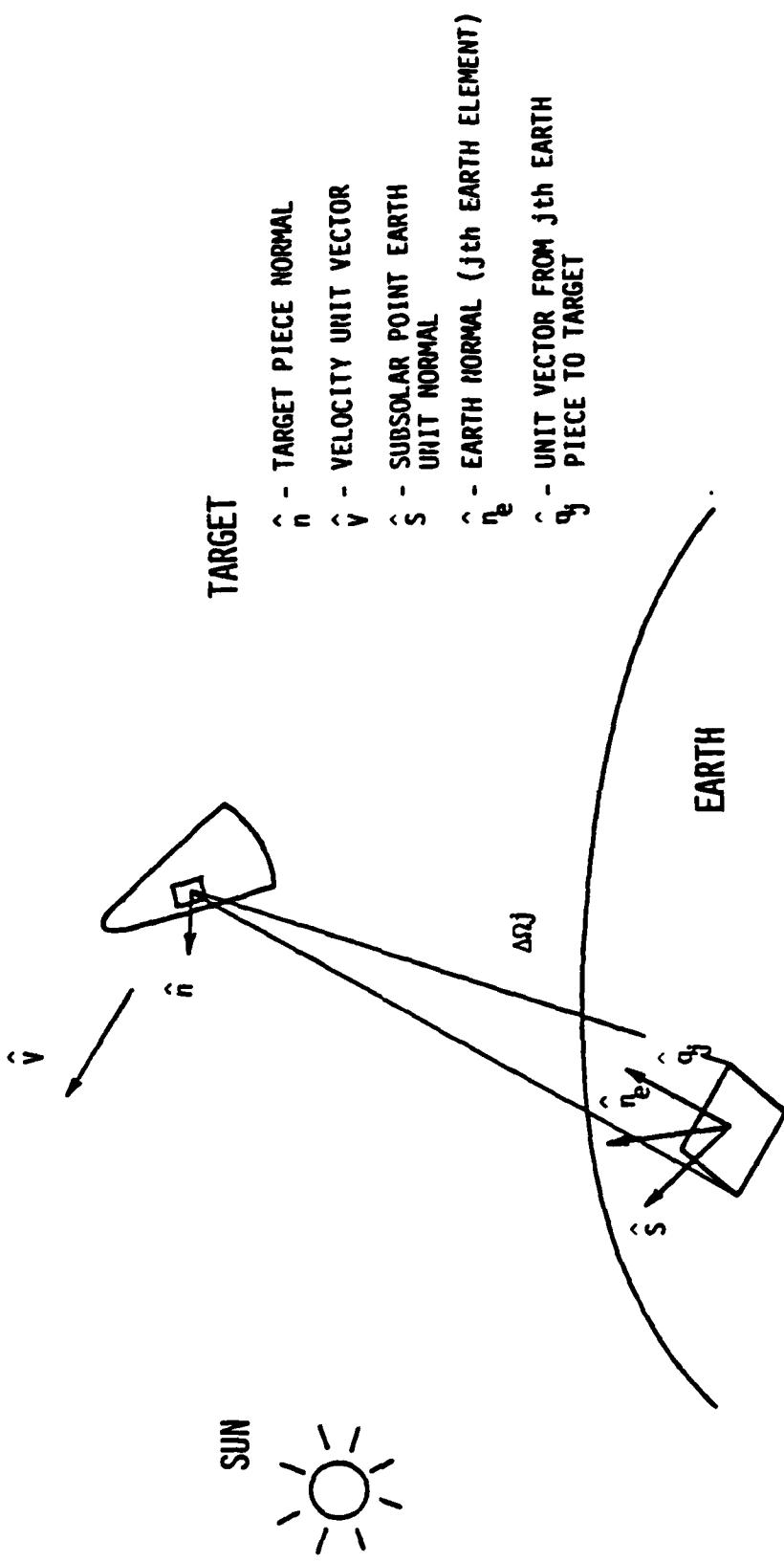


FIGURE 3-1. TARGET, SOLAR, EARTH GEOMETRY

TABLE 3-1. ATTITUDE AVERAGES FOR INCIDENCE FACTORS:  
 $G(\hat{n}, \hat{s}) = \hat{n} \cdot \hat{s} \theta(\hat{n} \cdot \hat{s})$

MODE*	$\langle G \rangle$	COMMENT
1	$G$	Not averaged. Instantaneous heat flux calculation
2	$\frac{1}{2\pi} \int_0^{2\pi} d\beta G$	Roll averaged
3	$\frac{1}{4\pi} \int d\Omega(\hat{s}) G = \frac{1}{4}$	$4\pi$ average: fast random tumbling

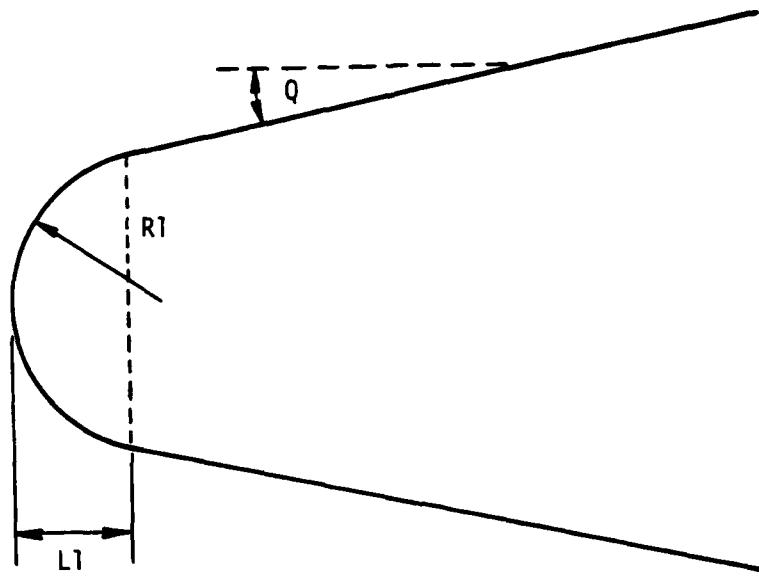
\*Input parameter

MBALL has the capability of treating transverse as well as longitudinal heat conduction as described below, so mode 1, instantaneous heating rate, is necessary to make use of this ability.

### 3.1.2 Internal Sources, Vehicle Geometry, Radiation Matrix

MBALL considers radiative coupling between the inner surfaces of a replica or balloon shape. It is possible to model a completely closed structure, or one in which the baseplate has been removed (see Section 3.3). No internal source of energy, such as batteries or wires, are included.

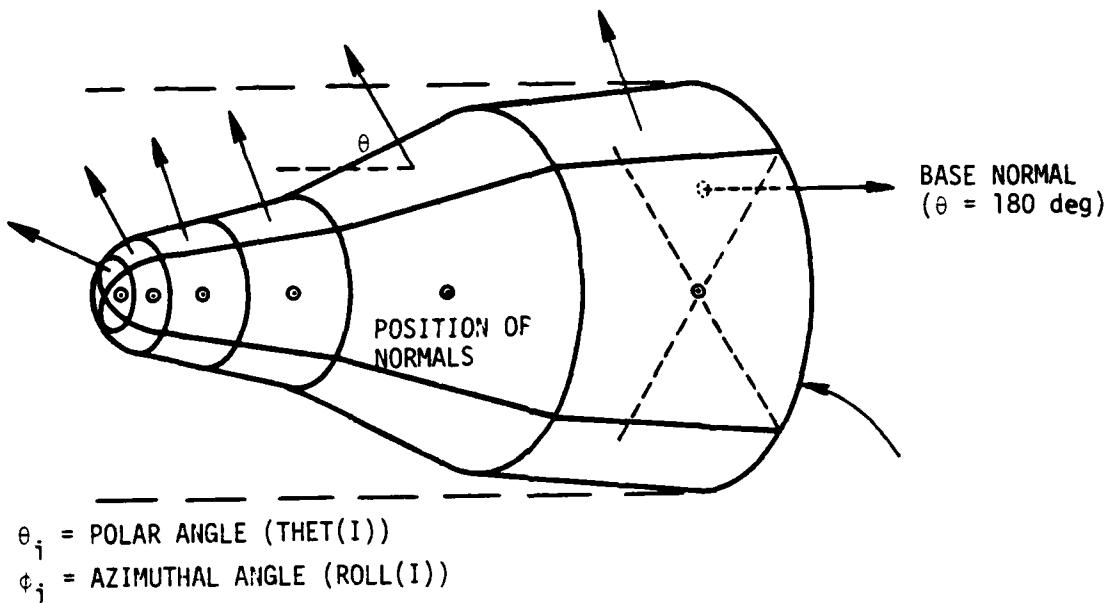
The user has the choice of modeling either a balloon (sphere) or an axially symmetric replica that can have a spherical nosecap, up to three frusta, and a flat baseplate. It is necessary to input the length of the nosecap such that it meets the first frustum tangentially (Figure 3-2). The surface area of the vehicle can be divided into as many as 50 pieces, called stations, where each station has an outer and inner surface normal, an initial temperature, an average thickness,



$R_1$  = RADIUS OF NOSECAP  
 $L_1$  = LENGTH OF NOSECAP  
 $Q$  = CONE ANGLE OF FIRST FRUSTUM  
 $L_1 = R_1 (1 - \sin Q)$

FIGURE 3-2. NOSECAP GEOMETRY

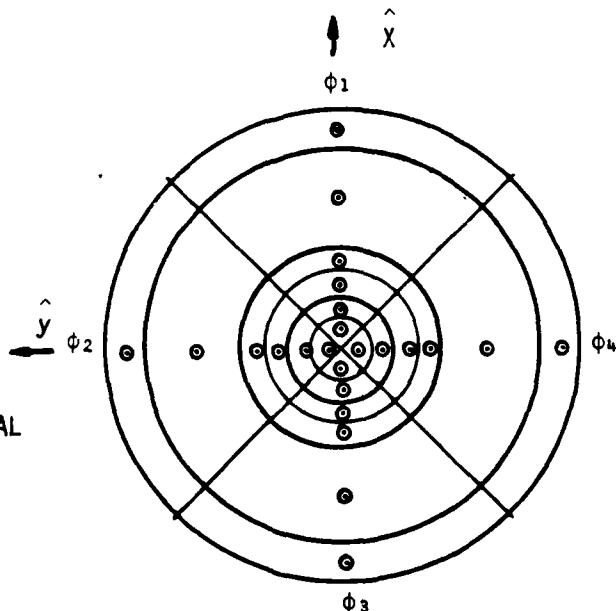
specific heat, thermal conductivity, solar absorptance, outside emissivity, inside emissivity, and density defined by the user. All stations on the nose section (nose, frustum, and base are sections) have the same areas, which are calculated by the program RVSNTH. The orientation of the outer normal of each station is defined by the angle it makes with the z axis (looking toward the nose) and the roll angle  $\phi$  (Figure 3-3). The running length,  $RL$ , distance from the nose along the axis z, and the distance from the axis,  $r$ , of the center of the station are calculated in RVSNTH (Figure 3-4). The polar and azimuthal angles are input by the user to EXOHEAT to determine the external heating rates. The polar angles for the normals on the frusta and base are chosen to agree with their respective slopes, but the choice for the polar angles on the spherical nose stations should be done such that all nose stations have equal areas (Figure 3-5). This ensures a consistent treatment of external and internal radiation in EXOHEAT and MBALL. The roll angle  $\phi$  is measured counterclockwise from the vehicle x axis, looking from the nose to the base (Figure 3-3).



THIS EXAMPLE HAS 7 LONGITUDINAL BY 4 AZIMUTHAL = 28 STATIONS TOTAL

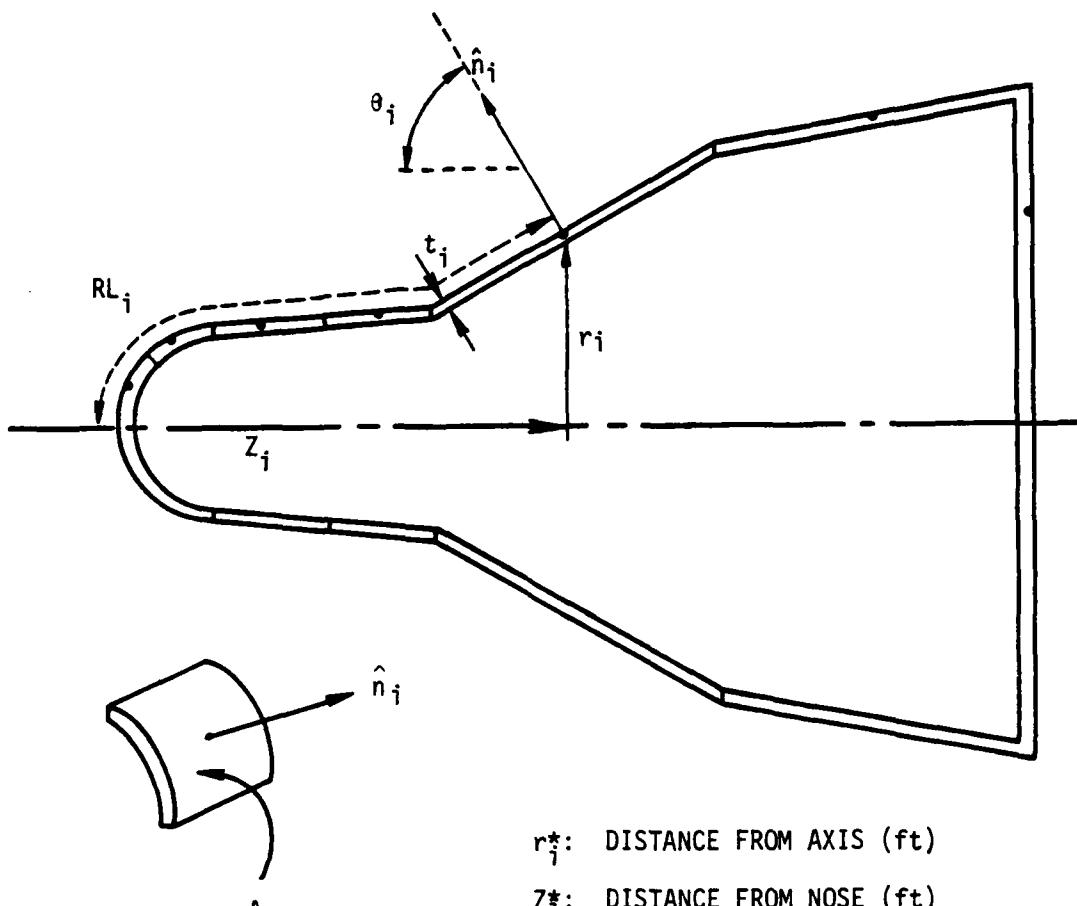
NUMBERING OF STATIONS  
 INCREASES LONGITUINALLY:

$\phi_1$ :	STATION 1-7
$\phi_2$	8-14
$\phi_3$	15-21
$\phi_4$	22-28



VEHICLE AS VIEWED FROM THE FRONT

FIGURE 3-3. STATION NORMALS (EXTERIOR)



$r_i^*$ : DISTANCE FROM AXIS (ft)

$z_i^*$ : DISTANCE FROM NOSE (ft)

$RL_i^*$ : RUNNING LENGTH (ft)

$A_i^*$ : STATION AREA ( $ft^2$ )

$t_i$ : THICKNESS (THK) (ft)

\*CALCULATED IN RVSNT

NOTE:  $\theta_i$  IS NOT PASSED TO EXOHEAT BY RVSNT, AND MUST BE CALCULATED BY THE USER FOR EXOHEAT INPUT, SEE FIGURE 3-5.

FIGURE 3-4. TARGET GEOMETRY

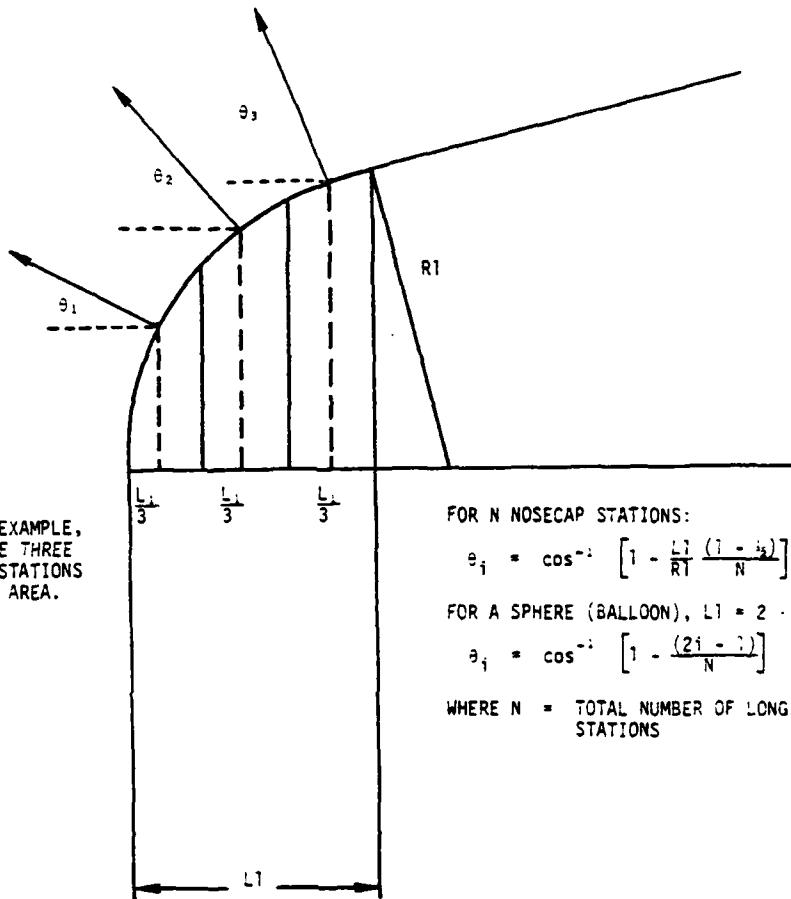


FIGURE 3-5. NOSECAP POLAR ANGLES

The stations can be distributed over the sections as desired, with the only restriction being that the number of roll angles times the number of polar angles be less than or equal to 50. Thus, it is possible to have 50 longitudinal stations along a single  $\phi$ , or 50 azimuthal stations at a single longitudinal position.

There is a small ambiguity in the term "station" as used in the RVSNT input. There, it means the number of stations on a given section, assuming only a single roll angle. The program automatically multiplies this by the number of roll angles to get the correct number of stations on the section.

The balloon is handled similarly to the replica, keeping in mind the criteria of equal areas for the determination of the polar angles.

Section 4 describes the input to RVSNT and EXOHEAT.

The radiation matrix,  $M_{ij}$ , will now be derived. The power,  $d^4P$ , incident on an area  $dA_j$  from a blackbody source of area  $dA_i$  can be written as (see Figure 3-6):

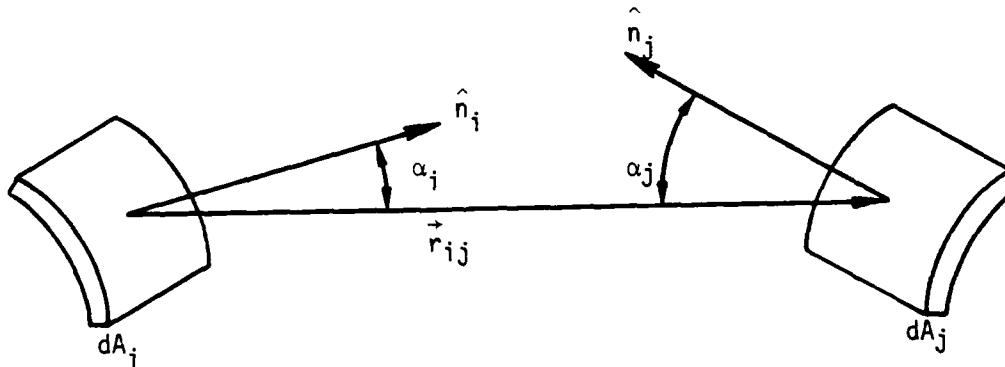


FIGURE 3-6. RADIATION EXCHANGE GEOMETRY

$$d^4P_j = \frac{\epsilon_j \sigma T_i^4}{\pi} \frac{\cos \alpha_i \cos \alpha_j}{|\vec{r}_{ij}|^2} dA_i dA_j \quad (3-13)$$

where

- $\alpha_i, \alpha_j$  - the angles the normals to  $dA_i, dA_j$  make with  $\hat{r}_{ij}$ , the vector pointing from  $dA_i$  to  $dA_j$
- $\epsilon_j$  - the emissivity of the piece  $j$
- $T_i$  - the temperature of the source
- $\sigma$  - the Stefan-Boltzmann constant.

The source surface over which Equation 3-13 is integrated is the interior surface of the replica or balloon model, which has been divided into  $N$  stations as described previously. Because the temperature and internal emissivity,  $\epsilon'$ , of each station are average values over the station, the integral of Equation 3-13 over the entire inner surface (excluding  $\vec{r}_{ij} = 0$ ) can be written as a sum over each station surface with  $\epsilon'$ ,  $T$  outside the integral, which results in:

$$\frac{d^2 P_j}{dA_j} = \epsilon_j \sigma \sum_i g_{ij}^{BB} T_i^4 \quad (3-14)$$

where  $g^{BB}$ , the blackbody radiation exchange matrix, is given by

$$g_{ij}^{BB} = \frac{1}{\pi} \int dA_i \cos \alpha_i \cos \alpha_j / |\vec{r}_{ij}|^2 . \quad (3-15)$$

To take into account the fact that the sources on the interior are not blackbodies, one replaces  $g^{BB}$  in Equation 3-14 by

$$g^T = [g^{BB^T} (1 - \rho' g^{BB^T}) \epsilon'] \quad (3-16)$$

where  $g$  is the true radiation exchange matrix and superscript  $T$  stands for "transposed". The reflectivity  $\rho'$  is given by  $\rho' = 1 - \epsilon'$ .

For a closed surface, one has

$$\sum_i g_{ij} = \sum_i g_{ij}^{BB} = 1. \quad (3-17)$$

Holes/open surfaces are correctly treated by setting  $\epsilon'_i = p'_i = 0$  for the surfaces transparent to radiation in the calculation of  $g$ . Equation 3-17 no longer applies to  $g$  if some surfaces are transparent.

The total power incident on  $A_j$  is approximated by

$$P_j = \epsilon'_j A_j \sigma \sum_i g_{ij} T_i^4. \quad (3-18)$$

The key to the calculation of  $g$  is, by Equation 3-16, the calculation of  $g_{ji}^{BB}$ . The calculation of  $g_{ji}^{BB}$  is accomplished by the subdivision of  $A_j$  into  $\eta$  longitudinal strips with center positions  $\vec{r}_{mj}$ , and a numerical integration is performed:

$$g_{ji}^{BB} = \frac{A_j}{\pi \eta} \sum_{m=1}^{\eta} \cos \alpha_{(j,m)} \cos \alpha_{(i,m)} / |\vec{r}_{mj}|^2. \quad (3-19)$$

To ensure that no serious errors result from this numerical procedure, Equation 3-17 is imposed on  $g^{BB}$ .

Excluding external sources for the moment, the total radiation flux into  $A_j$  is the amount incident from the interior minus the internal and external emission of  $A_j$ :

$$P_j \text{ TOTAL} = \left[ -\sigma(\epsilon_j + \epsilon'_j) T_j^4 + \sigma \sum_i \epsilon'_j T_i^4 g_{ij} \right] \cdot A_j \quad (3-20)$$

which can be written as:

$$P_j \text{ TOTAL} = \sum_i M_{ij} T_i^4 \quad (3-21)$$

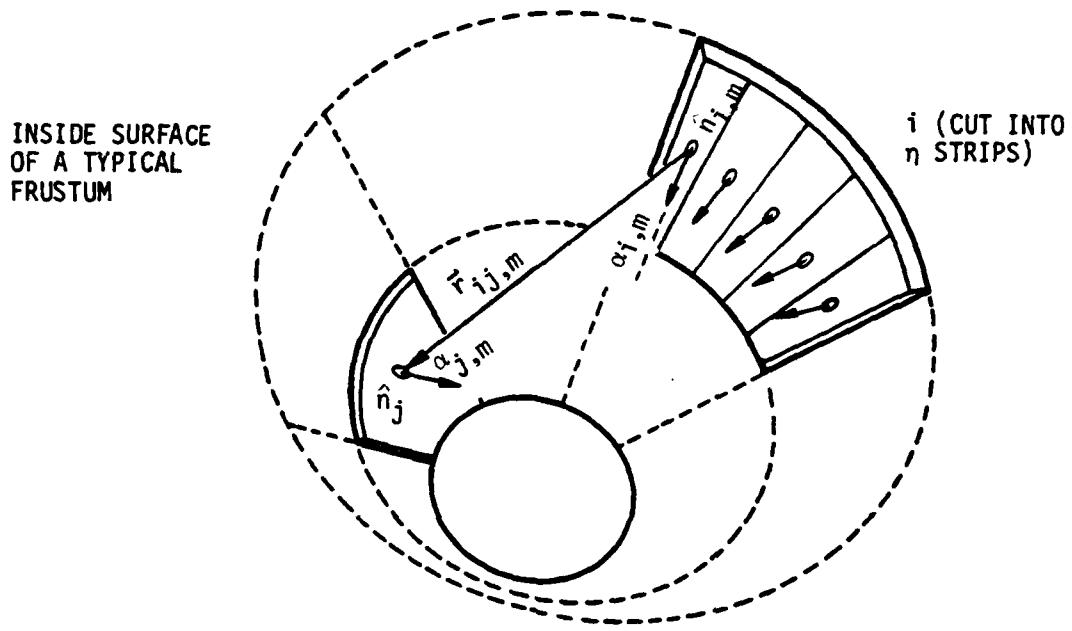


FIGURE 3-7. INTERNAL RADIATION GEOMETRY

where

$$M_{ij} = \left[ -\sigma(\epsilon_j + \epsilon_j') \delta_{ij} + \sigma \sum_i \epsilon'_j g_{ij} \right] \cdot A_j \quad (3-22)$$

with  $\delta_{ij} = \begin{cases} 1 & i=j \\ 0 & i \neq j \end{cases}$ .

The term  $g_{ii}$ , the "self contribution", can be understood as a correction to the energy radiated away internally by a curved piece. One edge of the piece radiates back toward the opposite edge, thus reducing the total outward energy flow.

### 3.1.3 Conduction Matrix

Equation 3-7 expressed the conduction integral of Equation 3-6 as a sum over adjoining areas of the jth piece:

$$\int_{A_j: \text{adjoining}} (\vec{k} \nabla T) \cdot \hat{\eta}_j dA_j + \sum_{\text{adjoining}} A_{\text{adjoining}} \cdot k_{\text{eff}} \frac{T_{\text{adjoining}} - T_j}{\Delta X_{j:\text{adjoining}}} \quad (3-23)$$

where all terms are defined below Equation 3-7. This sum can be expanded into its component terms by keeping in mind how the target surface has been divided:

$$\begin{aligned} \sum_{\text{adjoining}} A_{\text{adjoining}} k_{\text{eff}} \frac{T_{\text{adjoining}} - T_j}{\Delta X_{j:\text{adjoining}}} &= \frac{A_{(\ell-)} k_{\text{eff}}(j-\theta, j) (T_{j-\theta} - T_j)}{\Delta X_{j-\theta, j}} \\ &+ \frac{A_{(\ell+)} k_{\text{eff}}(j+\theta, j) (T_{j+\theta} - T_j)}{\Delta X_{j+\theta, j}} \\ &+ \frac{A_{(T-)} k_{\text{eff}}(j-\phi, j) (T_{j-\phi} - T_j)}{\Delta X_{j-\phi, j}} \\ &+ \frac{A_{(T+)} k_{\text{eff}}(j+\phi, j) (T_{j+\phi} - T_j)}{\Delta X_{j+\phi, j}} \end{aligned} \quad (3-24)$$

where  $j \pm \theta$ ,  $j \pm \phi$  label the pieces adjacent to the jth piece, in the longitudinal and azimuthal directions, respectively (see Figure 3-8). The remaining terms are defined as:

$$A_{(\ell \pm)} = \Delta\phi \left( \frac{r_j + r_{j \pm \theta}}{2} \right) \left( \frac{t_j + t_{j \pm \theta}}{2} \right) \quad (3-25a)$$

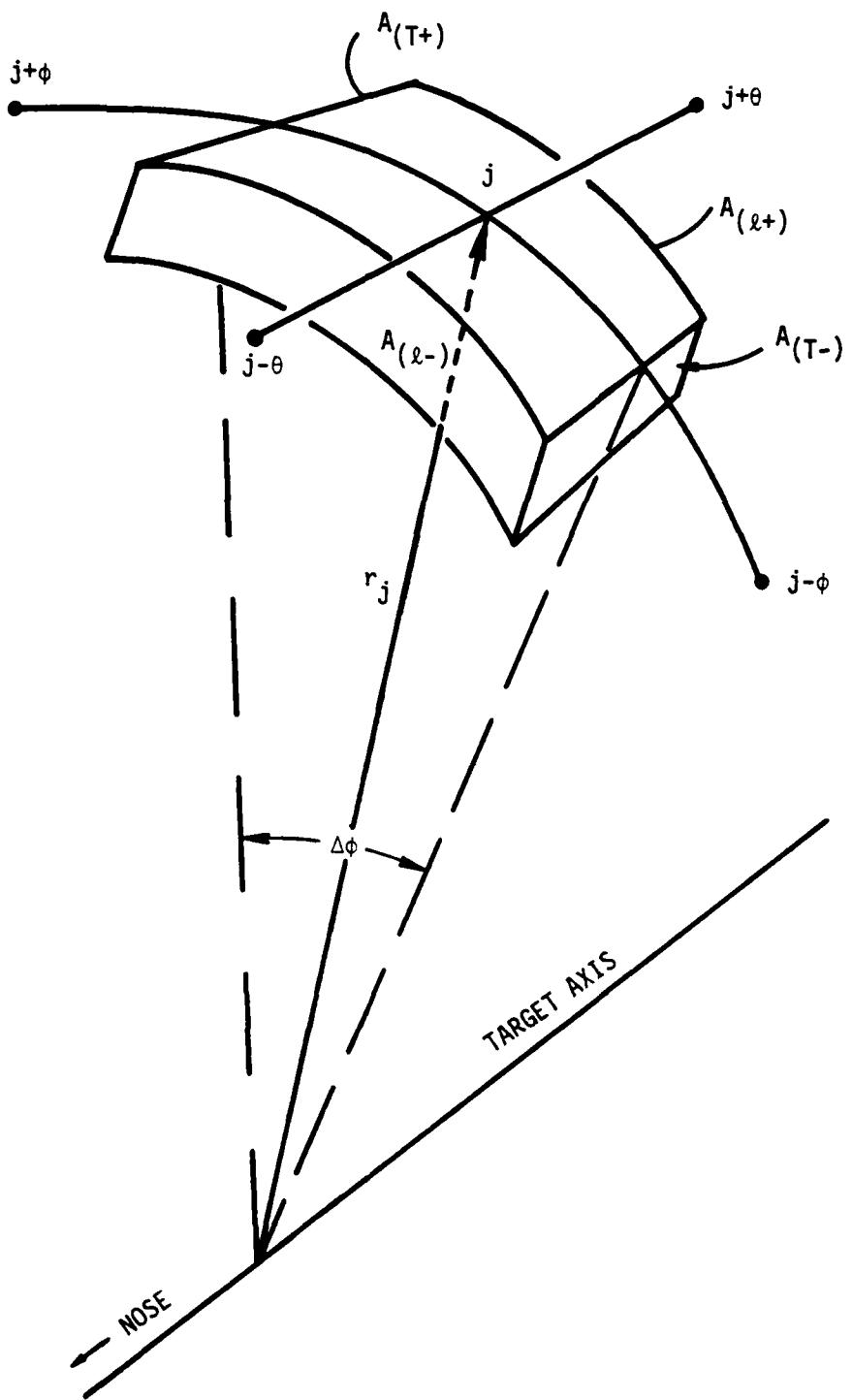


FIGURE 3-8. CONDUCTION GEOMETRY

where

$$\Delta\phi = \frac{1}{2} (\phi_j + \phi_{j+\theta}) - \frac{1}{2} (\phi_j + \phi_{j-\theta}) \quad (3-25b)$$

$$A_{(T\pm)} = \left[ \left( \frac{RL_{j+\theta} + RL_j}{2} \right) - \left( \frac{RL_j + RL_{j-\theta}}{2} \right) \right] \left( \frac{t_j + t_{j+\theta}}{2} \right) \quad (3-25c)$$

$$\Delta X_{j\pm\theta, j} = \pm (RL_{j\pm\theta} - RL_j) \quad (3-25d)$$

$$\Delta X_{j\pm\theta, j} = |\phi_{j\pm\theta} - \phi_j| \left( \frac{r_j + r_{j\pm\theta}}{2} \right) \quad (3-25e)$$

$$k_{eff}(n, \xi) = \frac{2}{\frac{1}{k_n} + \frac{1}{k_\xi}} \quad \text{where } n, \xi = j, j\pm\theta, j\pm\phi \quad (3-25f)$$

and where  $r$ ,  $\phi$ ,  $t$ ,  $RL$  are defined in Figure 3-4.

By defining longitudinal and transverse conduction coefficients of the form:

$$\alpha_{j\pm\theta} = \frac{A_{(\ell\pm)} k_{eff}(j\pm\theta, j)}{\Delta X_{j\pm\theta, j}} \quad (3-26a)$$

$$\beta_{j\pm\phi} = \frac{A_{(T\pm)} k_{eff}(j\pm\phi, j)}{\Delta X_{j\pm\phi, j}} \quad (3-26b)$$

Equation 3-24 can be written as:

$$\sum_{\text{adjoining}} A_{\text{adjoining}} k_{eff} \frac{T_{\text{adjoining}} - T_j}{\Delta X_{j:\text{adjoining}}} = \alpha_{j-\theta} T_{j-\theta} + \alpha_{j+\theta} T_{j+\theta} + \beta_{j-\phi} T_{j-\phi} + \beta_{j+\phi} T_{j+\phi} - T_j (\alpha_{j-\theta} + \alpha_{j+\theta} + \beta_{j-\phi} + \beta_{j+\phi}). \quad (3-27)$$

Thus, by defining a conduction matrix of the form

$$k_{ij} = \begin{bmatrix} -(a_{1+0} + a_{1-0} + a_{1-1}) & a_{1+0} & 0 & \dots & a_{1+0} & 0 & 0 & \dots & a_{1-0} & 0 & 0 & 0 \\ a_{1+0} & -(a_{1+0} + a_{2+0} + a_{2-0}) & a_{2+0} & \dots & 0 & a_{2+0} & 0 & \dots & 0 & a_{2-0} & 0 & 0 \\ 0 & a_{2+0} & -(a_{2+0} + a_{3+0} + a_{3-0}) & \dots & 0 & 0 & a_{3+0} & \dots & 0 & 0 & a_{3-0} & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \vdots \end{bmatrix}$$

(3-28)

The sum (including  $i = j$ )  $\sum k_{ij} T_i$  will contain all conduction terms associated with the  $j$ th piece, so that Equation 3-27 can be written as:

$$\sum_{\text{adjoining}} A_{\text{adjoining}} k_{\text{eff}} \frac{T_{\text{adjoining}} - T_j}{\Delta X_{j:\text{adjoining}}} = \sum_i K_{ij} T_i. \quad (3-29)$$

MBALL updates the  $K_{ij}$  with the changing temperatures of the stations through the thermal conductivity  $k_{\text{eff}}$ , which is, in general, a function of the temperature.

Because the end pieces, i.e., the first nose piece or the last baseplate piece, have only one longitudinal conduction contribution, the appropriate elements of the matrix  $K_{ij}$  are set to zero.

### 3.1.4 Thermal Mass

The left side of Equation 3-10 can be written as:

$$\rho_j V_j \frac{d}{dt} \int_{T_{0j}}^{T_j} c_{pj}(T'') dT'' = \int_{T_{0j}}^{T_j} m_j dT_j'' \quad (3-30)$$

where  $m_j = \rho_j V_j c_{pj}$  is defined as the thermal mass of the  $j$ th piece, and is, in general, a function of the temperature because it contains the specific heat. If the vehicle being modeled has a skin structure, i.e., if it is built up of layers of materials with different densities

and thermal properties, an appropriate average of these quantities must be made over all of the layers, because only a lumped thermal mass is treated in MBALL for each station. Equations 3-31 through 3-33 give the average density, specific heat, and thermal conductivity that are necessary:

$$\bar{\rho}_j = \sum_k \frac{\rho_{j,k} \Delta x_{j,k}}{\sum_k \Delta x_{j,k}} \quad (3-31)$$

$$\bar{c}_{pj} = \frac{\sum_k \rho_{j,k} \Delta x_{j,k} c_{pj,k}}{\sum_k \rho_{j,k} \Delta x_{j,k}} \quad (3-32)$$

$$\bar{k}_j = \frac{\sum_k \Delta x_{j,k} k_{j,k}}{\sum_k \Delta x_{j,k}} \quad (3-33)$$

where  $\Delta x_{j,k}$ ,  $\rho_{j,k}$ ,  $c_{pj,k}$ , and  $k_{j,k}$  are the thickness, density, specific heat, and thermal conductivity, respectively, of the  $k$ th layer of piece  $j$ . Because the specific heat and thermal conductivity are input by the user for different temperatures, an appropriate  $\bar{c}_{pj}$  and  $\bar{k}_j$  should be input for each temperature. The temperature calculated for the  $j$ th piece by MBALL is the average temperature over the thickness of the skin. Rapidly varying specific heats are accounted for in the solution of the heat equation described in the next section.

The option exists to set the thermal mass to zero (BALL = 2) for thin skins, in which case the temperature of each station is solved for assuming radiative equilibrium:

$$(\epsilon_j + \epsilon_j') \sigma T_j^4 = Q_{ext,j} + Q_{int,j} \quad (3-34)$$

where  $Q_{ext,j}$ ,  $Q_{int,j}$  are the external and internal power per unit area, respectively, incident on the  $j$ th piece.

### 3.2 SOLUTION OF THE HEAT EQUATION

#### 3.2.1 Linearization

For numerical calculations, it is necessary to linearize both the left- and right-hand side of Equation 3-10. Proceeding with the left-hand side first:

$$\frac{\partial}{\partial t} \int_{T_{0j}}^{T_j} m_j(T_j'') dT_j'' \rightarrow \frac{m_j(T_j')(T_j - T_j')}{\Delta t} + \frac{1}{\Delta t} \int_{T_{0j}}^{T_j'} m_j(T'') dT'' \quad (3-35)$$

where  $\Delta t$  is the time step,  $T_{0j}$  is the initial temperature of piece  $j$ , and  $T_j'$  is an estimated value for the new temperature  $T_j$  after the time  $\Delta t$ . The determination of  $\Delta t$  and  $T'$  is discussed below. Because the temperature change is being sought over a finite interval  $\Delta t$ , it is appropriate to replace the instantaneous value of the right-hand side of Equation 3-10 by a value averaged over the interval:

$$Q_{ext,j}(t) A_j + \sum_i (M_{ij} T_i^4 + K_{ij} T_i) \rightarrow Q_{ext,j}(t + \frac{\Delta t}{2}) A_j + \sum_i (M_{ij} \bar{T}_i^4 + K_{ij} \bar{T}_i) \quad (3-36)$$

where  $\bar{T}_i$ , the average temperature, is given by  $\bar{T}_i = 1/2 (T_{oi} + T_i)$ , with  $T_{oi}$ ,  $T_i$  the old and new temperature of the  $i$ th piece, respectively, and the external flux  $Q_{ext,j}$  is interpolated to the midpoint of the time interval. The  $\bar{T}_i^4$  term can be linearized in the following way: expanding  $T_i^4$  in a Taylor's expansion about an estimated final temperature  $T_i'$ :

$$T_i^4 = T_i'^4 + 4 T_i'^3 (T_i - T_i') \quad (3-37)$$

and substituting this expression for  $T_i$  in  $\bar{T}_i^4$  yields (after a further approximation):

$$\bar{T}_i^4 \rightarrow \left[ \frac{1}{2} (T_i' + T_{oi}) \right]^4 + 2 \left[ \frac{1}{2} (T_i' + T_{oi}) \right]^3 \cdot (T_i - T_i') .$$

(3-38)

Defining the term  $\bar{T}_i' = \frac{1}{2} (T_i' + T_{oi})$  for convenience, the full linearized heat equation can be written:

$$\begin{aligned} \frac{m_j (T_j') (T_j - T_j')}{\Delta t} + \frac{1}{\Delta t} \int_{T_{oj}}^{T_j'} m_j(T'') dT'' &= Q_{ext,j} \left( t + \frac{\Delta t}{2} \right) A_j \\ &+ \sum_i \left\{ M_{ij} \left[ \bar{T}_i'^4 + 2 \bar{T}_i'^3 (T_i - T_i') \right] \right. \\ &\quad \left. + \frac{k_{ij}}{2} (T_{oi} + T_i) \right\} \end{aligned} \quad (3-39)$$

If the terms in Equation 3-39 are grouped to isolate  $T_j$ , the following equation results:

$$\sum_i c_{ij} T_i = B_j \quad (3-40)$$

where

$$c_{ij} = \begin{cases} \frac{m_j (T_j')}{\Delta t} - 2 M_{jj} \bar{T}_j'^3 - \frac{1}{2} k_{jj} & \text{for } i = j \\ - (2 M_{ij} \bar{T}_i'^3 + \frac{1}{2} k_{ij}) & \text{for } i \neq j \end{cases} \quad (3-41)$$

and

$$B_j = Q_{ext,j} \left( t + \frac{\Delta t}{2} \right) A_j + \frac{m_j(T_j') T_j'}{\Delta t} - \frac{1}{\Delta t} \int_{T_{0j}}^{T_j'} m_j dT \\ + \sum_i \left[ M_{ij} (\bar{T}_i^4 - 2\bar{T}_i^3 \cdot T_i') + \frac{K_{ij}}{2} T_{oi} \right].$$

Equation 3-40 is solved for the  $T_j$  by matrix inversion, and if the  $T_j$  are within one-half of one percent of the estimated  $T_j'$ , the  $T_j$  are the new temperatures. Otherwise, the  $T_j$  become the estimated temperatures  $T_j'$  and Equation 3-40 is evaluated again for  $T_j$ . This process is repeated until the  $T_j$  are found.

### 3.2.2 Estimation of $\Delta t$ , $T'$

It is important that the diagonals of the matrix  $c_{ij}$  be greater than zero, so that the time development is stable. Choosing the time step,  $\Delta t$ , such that

$$\Delta t < \frac{m_j (T_{0j})}{2M_{jj} T_{0j}^3 + 1/2 K_{jj}} \quad (3-43)$$

guarantees that the  $c_{ii}$  are all positive.

The estimated temperature,  $T_j'$ , is found by solving the heat equation with the right-hand side approximated by an expansion about the initial temperature  $T_{0j}$ :

$$\int_{T_{0j}}^{T_j'} m_j dT = \Delta t \left\{ \sum_i \left[ M_{ij} T_{oi}^4 + K_{ij} T_{oi} + Q \left( t + \frac{\Delta t}{2} \right) A_j \right] \right. \\ \left. + (4 M_{jj} T_{0j}^3 + K_{jj}) (T_j' - T_{0j}) \right\}. \quad (3-44)$$

$T_j'$  is then used to compute the specific heat at  $T_j'$ ,  $c_{pj}(T_j')$ , and the time step is checked to ensure that

$$\Delta t < \frac{m_j(T_j')}{4 M_{jj} T_{oj}^3 + K_{jj}} . \quad (3-45)$$

If the time step does not satisfy Equation 3-45, a smaller step is chosen, and a new value for  $T_j'$  is found from Equation 3-44. This process is repeated until  $\Delta t$  is found to satisfy Equation 3-45. The time between trajectory points at which the temperatures are desired is used as the time step if this time is smaller than that defined by Equation 3-45.

### 3.3 OPEN SURFACE OPTION

It is possible to remove any station (more than one station may be removed) from the replica or balloon shape to simulate an open surface (hole) at that station. This is accomplished by setting the initial temperature of the station equal to 0°R on material property card 6.1 (Table 4-3). The conductivity and radiative coupling of any open station to the rest of the vehicle is set equal to zero. Any external fluxes entering the vehicle through the open station are neglected, however, so the percentage of open surface to total vehicle surface should be small to minimize the error in determining the true flux on an interior surface. Thus the total number of removed stations should be small.

An average temperature is assigned to an open surface for signature calculations in BIDIREC. This average temperature is representative of the internal energy that is passing through the hole from the interior. The average temperature of the missing station,  $\bar{T}_{open}$ , is found by evaluating:

$$\bar{\epsilon} \bar{T}_{open}^4 = \sum_i \epsilon_i' g_{i,open} T_i^4 \quad (3-46)$$

Because this is an approximation, the internal emissivities,  $\epsilon_i'$ , are assumed to be about the same, and they cancel with the average emissivity,  $\bar{\epsilon}$ . The  $g_{i,open}$  term is the geometric factor for the  $i$ th surface as seen by the open surface.

## 4. INPUT SPECIFICATIONS

### 4.1 DISC UTILIZATION

MBALL requires the disc files used in EXOHEAT for its operation. Table 4-1 summarizes the necessary tapes and their utilization.

The thermophysical data (15) may be input by the user. The trajectory data (7) and earthshine data (4) must be input in the BASIC option. Note, however, that the user can supply trajectory data to tape 7 by using the trajectory card input option (Ref. 1).

TABLE 4-1. EXOHEAT (WITH MBALL) DISC FILES

DEVICE/ TAPE	UTILIZATION	ORIGIN
4	Earthshine Data	OSC Data Base
7	Trajectory Data	BALLIS/BASIC
15	Thermophysical Data	OSC Data Base
16	Temperature	MBALL
23	Vehicle Geometry	RVSNTB/BASIC

### 4.2 CARD INPUT

The MBALL card input consists of two parts: RVSNTB input and EXOHEAT input.

RVSNTB calculates the necessary geometrical parameters for MBALL from a modest user input (for both replica and balloon shapes). The user must be careful, however, to enter the nosecap polar angles [THET(I)] in the EXOHEAT input that are calculated by RVSNTB (see Figure 3-5) because RVSNTB does not pass them to EXOHEAT. RVSNTB input is displayed in Table 4-2.

TABLE 4-2. RVSNT INPUT SUMMARY

CARD COLUMN	VARIABLE	FORMAT	UNITS	DESCRIPTION
FIRST CARD: TITLE (FORMAT 3A4)				
1-12	(NTI(I), I = 1,3)	3A4		RV title
SECOND CARD: UNIT DESIGNATOR (FORMAT I3)				
1-3	FLAG	I3		Designates what units of length the input is in: = 1, ft = 3, cm = 2, in. = 4, m
THIRD CARD: NOSECAP (FORMAT I5, 5X, 2F10.4, I5)				
1-5	N	I5		Number of stations on nose cap
11-20	R1	F10.4	(FLAG)	Nosecap radius (or balloon radius)
21-30	L1	F10.4	(FLAG)	Length of nosecap*† (or balloon diameter)
31-35	KROL	I5		Number of azimuthal divisions
FOURTH CARD: FRUSTA CONTROL (FORMAT I5, 5X, F10.4)				
1-5	MN	I5		Number of frusta (3 maximum)
11-20	Q	F10.4	deg	Cone angle of first frustum
FRUSTRA CARDS: (ONE PER FRUSTUM) (FORMAT I5, 5X, 2F10.4)				
1-5	N	I5		Number of stations on frustum
11-20	Q	F10.4	deg	Cone angle
21-30	L1	F10.4	(FLAG)	Length of frustum
BASE CARD (FORMAT F5.0)				
1-5	AN	F5.0		Number of stations on base

\*See Figure 3-2 for replica.

†If balloon shape is desired, L1 = 2 × R1 and the third card is the last RVSNT card.

The first card contains the user's title (up to 12 letters). The second card defines the units in which the target dimensions are input (these are changed internally to feet for MBALL). The nosecap information is input on the third card: N, the number of stations refers to the number of longitudinal stations along a single running length of the nosecap. The total number of nosecap stations is KROL X N. For a replica shape, the nosecap must fit tangentially to the first frustum, so that L1 is computed by the user according to Figure 3-2. If a balloon shape is desired (i.e., sphere), the user should set L1 equal to twice R1. RVSNT bypasses the frustum calculations when  $L1 = 2 \times R1$ , so the nosecap card is the last RVSNT card when a balloon shape is desired. For the replica shape, the next card sets the number of frusta and defines the first cone angle, and this card is followed by a card for each frustum. Again, the total number of stations on each frustum is  $N \times KROL$ . Finally, the base card contains the number of longitudinal stations on the base (in F format), to be multiplied by KROL to get the total number of base stations.

EXOHEAT input is shown in Table 4-3. The first card is the NAMELIST/SOLAR data. This defines the earthshine mission map. The format for this card is (note the leading blank denoted by the b):

```
b$SOLAR ITYP = ...,$
```

with the variables separated by commas. A \$ ends the namelist.

The next card set defines all station normals by their azimuthal and polar angles. The polar angles, THET(I), should agree with those calculated by RVSNT (see Figure 3-5).

Card 3.1 determines the external flux averaging mode (see Table 3-1). MODE = 1 allows the full use of MBALL's capability for azimuthal, as well as longitudinal, heat conduction. In MODE = 2 (roll average), all azimuthal stations experience the same roll averaged external fluxes. Azimuthal conduction can still be important in this case, however, if adjacent stations have different thermal masses or thermal properties.

TABLE 4-3. EXOHEAT (WITH MBALL) CARD INPUT SUMMARY

NATURAL ENVIRONMENT SPECIFICATION	
NAMELIST/SOLAR	
VARIABLE	DESCRIPTION
ITYPE	Sun position option = 1 - Subsolar latitude and longitude (SLAT, SLON) input = 2 - Subsolar point calculated from MONTH, IDAY, HOUR, where HOUR is GMT time = 3 - Subsolar point calculated from MONTH, IDAY, HOUR, ELG, where HOUR is local Sun time at longitude ELG
SLAT	Sun latitude (deg)
SLON	Sun longitude
MONTH	Month of year - 1 to 12
IDAY	Day of month - 1 to 30
HOUR	24 hr time (1:15 p.m. ~ 13.25)
ELG	Reference longitude for local time (deg)
IPRINT	Earthshine map printing option = 0 - Do not print earthshine radiances ≠ 0 - Print earthshine map
ICLD	Cloud cover option = 0,1 - Average seasonal geographic earthshine = 2 - Cloudy radiance reduction = 3 - Clear radiance enhancement = 4 - Statistical - random geographical variation between clear and cloudy = 5* - Clear and cloudy sections are positioned over the Earth = 6* - Same as ICLD = 5, but different albedos are used for water, land, vegetation, snow, and ice on the Earth's surface.

\*See Table 4-20, Reference 1.

TABLE 4-3 - Continued

BODY SHAPE			
CARD	PARAMETER	FORMAT	DESCRIPTION
2.1	KROL	I5	Numbers of azimuthal divisions
2.2	(ROLL(I), I=1, KROL)	8F10.5	Azimuthal angle (deg)
2.3	KTHET	I5	Number of polar angles
2.4	(THET(I), I=1, KTHET)	8F10.5	Polar angles (deg)*
FLUX AVERAGING MODE			
3.1	MODE	I5	= 1 - Instantaneous = 2 - Roll averaged fluxes = 3 - Spherical averaged fluxes
BALLOON OPTION			
NAMELIST/IBALL			
VARIABLE	DESCRIPTION		
BALL <sup>†</sup>	= 2 - Radiative equilibrium = 3 - Thermal mass used		
CONDUCTION SPECIFICATION			
CARD	PARAMETER	FORMAT	DESCRIPTION
5.1	COND  FLAG	(5X,L5, F5.2)	T = Conduction F = No conduction 0 = Thermal prop. updated with time 1 = Thermal prop. not updated

\*Nosecap THET(I) should agree with polar angles calculated in RVSNTM (see Figure 3-5).

<sup>†</sup>BALL must equal 2 or 3 for MBALL to be called.

TABLE 4-3 - Concluded

MATERIAL PROPERTY SPECIFICATION			
(One card for each station)		The station index for the Mth roll angle and Nth polar angle is $I = (M-1) \cdot KTHET + N$	
CARD	PARAMETER	FORMAT	DESCRIPTION
6.1	JMAT	(I10, 2F10.5)	Material identification code word >200 - data taken from OSC data base <200 data taken from card sets composed of card types 7.1 and 7.2
	THK		Material thickness (ft)
	TOLD		Initial temperature* ( $^{\circ}$ R)
If some of JMATS are less than 200 material property data will be input on these cards			
7.1	NCP	(I5, 7F10.2)	Number of temperatures and number of 7.2 cards
	DENS		Density ( $\text{lb}/\text{ft}^2$ )
	E0		Outside emissivity
	EI		Inside emissivity
	AL		Absorptance
7.2	TEMP	(3E12.5)	Temperature ( $^{\circ}$ R)
	CP		Specific heat ( $\text{Btu}/\text{lb}/^{\circ}\text{R}$ )
	TCON		Thermal conductivity ( $\text{Btu}/\text{ft}/^{\circ}\text{R}/\text{sec}$ )

\*TOLD = 0 for an open surface

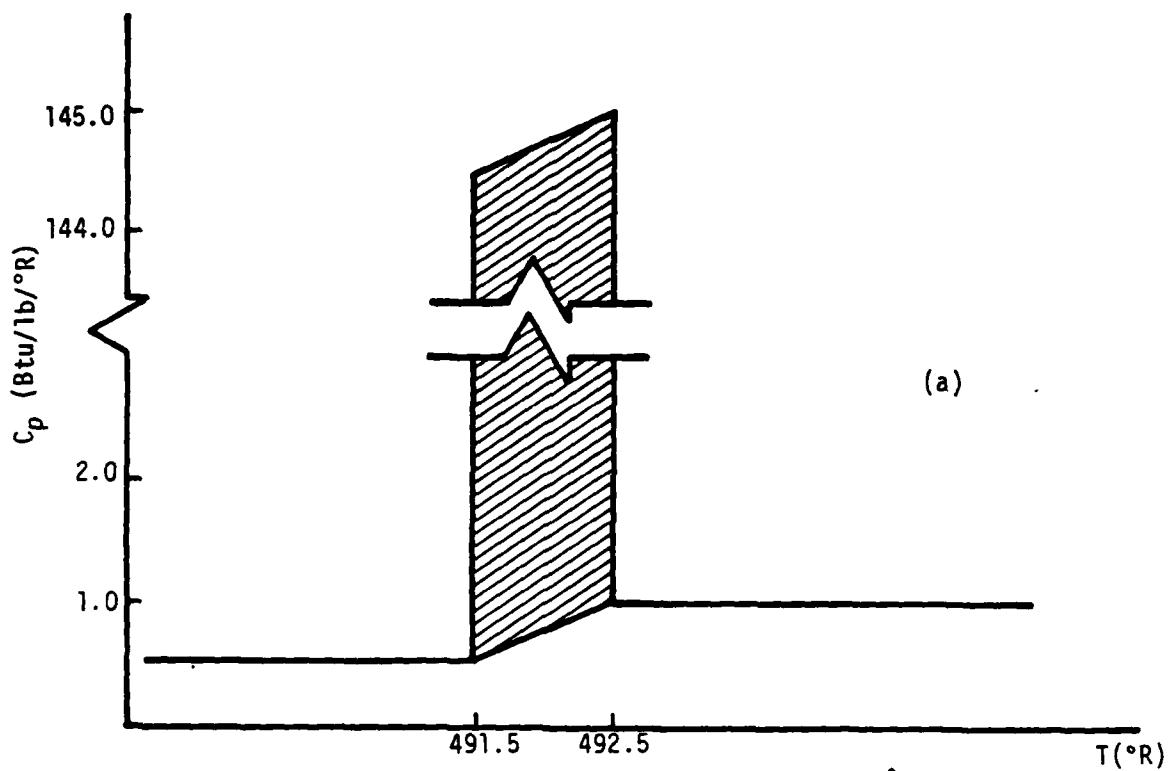
The next card is NAMELIST/IBALL input. If BALL = 2, a zero thermal mass is assumed for all of the stations, and Equation 3-34 is solved for the temperatures. BALL = 3 uses the complete thermal mass solution of Section 3. BALL must equal two or three for MBALL to be called. The format for this namelist is:

```
b$IBALL    BALL = 2(or 3)      $
```

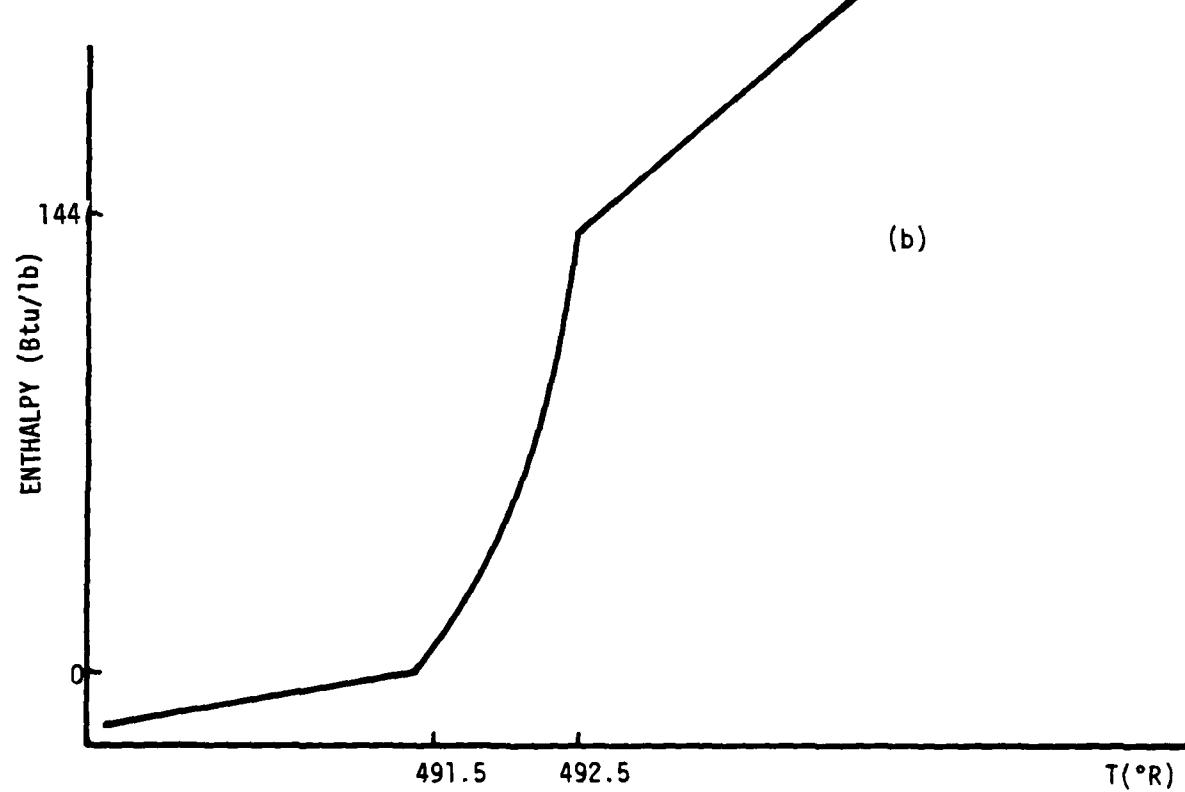
Card 5.1 chooses the two options: conduction on/off, and thermal properties updated/not updated with time.

Card set 6 defines the material, thickness, and initial temperature of each station (one card per station). An open station is flagged by setting its initial temperature equal to 0. If JMAT is greater than 200, the thermophysical properties are taken from the OSC data base. Table 4-4 contains the OSC thermophysical property code words. If JMAT is less than or equal to 200, the thermophysical properties must follow card set 6. If a station is built up of layers of different materials, an appropriate average of their thermophysical properties over the thickness of the skin should be input in card set 7 (see Equations 3-31 through 3-33). Card set 7 is read whenever a JMAT less than or equal to 200 is encountered that is different from the previous station's JMAT.

To model a phase change, the specific heat can be thought of as a spiked function of the temperature, as is shown in Figure 4-1(a). The enthalpy, or energy content per unit mass of the station material, is the area under curve (a), and is shown in (b). The area of the shaded spike of curve (a) corresponds to the heat of fusion or heat of vaporization of a unit mass of material. In the case of water, this area would correspond to about 144 Btu/lb for the heat of fusion. Table 4-5 is a tabulation of the curve of Figure 4-1(a). Note that the width of the spike was chosen to be 1°R, but a smaller width may be chosen if the  $c_p$  is increased so that the area under the spike remains 144 Btu/lb.



(a)



(b)

FIGURE 4-1. SPECIFIC HEAT AND ENTHALPY OF WATER

TABLE 4-4. OSC THERMOPHYSICAL PROPERTY CODE WORDS

CODE WORD	MATERIAL
201	Carbon phenolic
202	Graphite
203	Silica phenolic, asbestos phenolic
204	Fused silica
205	Teflon
206	Porous stainless steel
207	Aluminum
208	Beryllium

TABLE 4-5. MODELING THE HEAT OF FUSION OF WATER

TEMPERATURE ( $^{\circ}$ R)	$c_p$ (Btu/lb/ $^{\circ}$ R)
480.00	0.5 (approx.)
491.49	0.5 (approx.)
491.50	144.5
492.50	145.0
492.51	1.0 (approx.)
500.00	1.0 (approx.)

#### 4.3 ACCESSING MBALL IN THE BASIC OPTION

MBALL is accessed in the BASIC option (note that Reference 1 and 3 do not contain MBALL) if the BASIC parameters HEATRV = EXOHEAT and TARGSYN = YES (this calls RVSNT). References 1 and 3 state that RVSNT should not be used with EXOHEAT, but this has been changed with the addition of MBALL. To compute signatures from the MBALL temperatures, input is necessary for SELECT and BIDIREC, because the modified RVSNT program does not compute the necessary parameters to these programs. Reference 1 describes SELECT and BIDIREC input.

## 5. EXAMPLES

### 5.1 WATER-JACKETED BALLOON

The first example (Figure 5-1) demonstrates the modeling of a water-jacketed balloon 1 m in radius consisting of an outer and inner skin of 1/32-in. teflon, supporting a 1/8-in. layer of water. It is necessary to average the density, specific heats, and thermal conductivities of the water and teflon over the layers of the balloon. The balloon's surface is divided into 36 stations of equal area. Table 5-1 shows the values chosen to represent the specific heat of water to model the phase change at 492 °R, and the specific heat of teflon. Also shown are the respective thermal conductivities. The averages used in the inputs were determined from Equations 3-31 through 3-33.

Trajectory cards have been input to place the balloon over the north pole with the axis of the balloon parallel to the axis of the Earth. The Sun is at 0° longitude and 0° latitude. The initial temperature of the entire balloon is 500 °R. Even though SELECT and BIDIREC input is present, the output from these programs is suppressed.

The MBALL output (Figure 5-2), following the average fluxes from the Sun, Earth, and molecular heating from EXOHEAT, is as follows for each station (units are in feet): the station area, perpendicular and parallel components of the station normal with respect to the vehicle axis, distance from the axis ( $r$ ), distance along the axis ( $z$ ), running length, and thermal mass (Btu/°R). The terms RAD and COND give an estimate of the initial radiation and conduction flow, respectively, involving the particular station. TMTP is the estimated time step  $\Delta t$  given by dividing the thermal mass TMASS by the sum of RAD and COND (this is one-half the  $\Delta t$  of Equation 3-43). ALF, EOUT, and EIN are the solar absorptivity and outside and inside emissivities, respectively. The temperatures (°R) are then output for each station.

CARD	1	11	21	31	41	51	61	71
1	BASIC							
2	TF2J	CARDS	SLATE	SOLAT	MATER	CALC	SIGMA	CALC
3	TFGL_YN	VCS						
4	7							
5	5.	9.						
6	11.	15.						
7	15.	24.						
8	31							
10	6.	3261.771	3.335.021	3281.8	2000.	90.		
11	12.	3261.771	3.335.021	3281.8	2000.	90.		
12	18.	3261.771	3.335.021	3281.8	2000.	90.		
13	24.	3261.771	3.335.021	3281.8	2000.	90.		
14	30.	3261.771	3.335.021	3281.8	2000.	90.		
15	36.	3261.771	3.335.021	3281.8	2000.	90.		
16	42.	3261.771	3.335.021	3281.8	2000.	90.		
17	48.	3261.771	3.335.021	3281.8	2000.	90.		
18	54.	3261.771	3.335.021	3281.8	2000.	90.		
19	60.	3261.771	3.335.021	3281.8	2000.	90.		
21	66.	3261.771	3.335.021	3281.8	2000.	90.		
21	72.	3261.771	3.335.021	3281.8	2000.	90.		
22	78.	3261.771	3.335.021	3281.8	2000.	90.		
23	84.	3261.771	3.335.021	3281.8	2000.	90.		
24	90.	3261.771	3.335.021	3281.8	2000.	90.		
25	96.	3261.771	3.335.021	3281.8	2000.	90.		
26	102.	3261.771	3.335.021	3281.8	2000.	90.		
27	108.	3261.771	3.335.021	3281.8	2000.	90.		
28	114.	3261.771	3.335.021	3281.8	2000.	90.		
29	120.	3261.771	3.335.021	3281.8	2000.	90.		
30	126.	3261.771	3.335.021	3281.8	2000.	90.		
31	132.	3261.771	3.335.021	3281.8	2000.	90.		
32	138.	3261.771	3.335.021	3281.8	2000.	90.		
33	144.	3261.771	3.335.021	3281.8	2000.	90.		
34	150.	3261.771	3.335.021	3281.8	2000.	90.		
35	156.	3261.771	3.335.021	3281.8	2000.	90.		
36	162.	3261.771	3.335.021	3281.8	2000.	90.		
37	168.	3261.771	3.335.021	3281.8	2000.	90.		
38	174.	3261.771	3.335.021	3281.8	2000.	90.		
39	180.	3261.771	3.335.021	3281.8	2000.	90.		
40	MEALL							
41	3							
42	6	100.	200.	6				
43	ICOLAF	ITYPE=1,SLAT=	,SLUN=	,MONTH=9,IDATE=15,IPRINT=4,ICLJ=3	\$			
44	?							
45	3.	60.	100.	100.	240.	300.		
46	6.							
47	37.56	60.	0.04	94.52	120.	146.44		
48	1							
49	SI(BALL BALL=3..?							
50	F	T						

CARD	1	11	21	31	41	51	61	71

FIGURE 5-1. BALLOON EXAMPLE INPUT

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CAFID	1	11	21	31	41	51	61	71
51		190 .0156	500.					
52		190 .0156	500.					
53		190 .0156	500.					
54		190 .0156	500.					
55		190 .0156	500.					
56		190 .0156	500.					
57		190 .0156	500.					
58		190 .0156	500.					
59		190 .0156	500.					
60		190 .0156	500.					
61		190 .0156	500.					
62		190 .0156	500.					
63		190 .0156	500.					
64		190 .0156	500.					
65		190 .0156	500.					
66		190 .0156	500.					
67		190 .0156	500.					
68		190 .0156	500.					
69		190 .0156	500.					
70		190 .0156	500.					
71		190 .0156	500.					
72		190 .0156	500.					
73		190 .0156	500.					
74		190 .0156	500.					
75		190 .0156	500.					
76		190 .0156	500.					
77		190 .0156	500.					
78		190 .0156	500.					
79		190 .0156	500.					
80		190 .0156	500.					
81		190 .0156	500.					
82		190 .0156	500.					
83		190 .0156	500.					
84		190 .0156	500.					
85		190 .0156	500.					
86		190 .0156	500.					
87	6	86.6	.9	.9	.2			
88		491.00	.353	.132				
89		491.49	.362	.165				
90		491.50	38.791	.165				
91		492.50	39.030	.165				
92		492.51	.530	.165				
93		500.00	.610	.167				
94	1	1	1	0				
95	FJ3098	11						
96	FQ30340	11						
97	6	1	1	1	2	1		
98	3	1	4	1	3			
99		1	5	1	7	1		
100		8	1	9	1	10		

CAFID	1	11	21	31	41	51	61	71
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FIGURE 5-1 - Continued

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C#FD	1	11	21	31	41	51	61	71
101		1	11	1	12	1		
103		13	1	14	1	15		
104		1	15	1	17	1		
105		18	1	19	1	20		
106		1	21	1	22	1		
107		23	1	24	1	25		
108		1	26	1	27	1		
109		28	1	29	1	30		
110		1	31	1	32	1		
111		33	1	34	1	35		
112	5	.349	24.09	0.	43.19	60.		
113	5	.349	55.35	0.	22.34	60.		
114	5	.349	61.25	0.	13.47	60.		
115	5	.349	99.74	0.	13.47	60.		
116	5	.349	12.64	1.	22.34	60.		
117	5	.349	155.91	0.	43.19	60.		
118	5	.349	24.09	60.	43.19	60.		
119	5	.349	52.35	60.	22.34	60.		
120	5	.349	61.25	60.	13.47	60.		
121	5	.349	99.74	60.	13.47	60.		
122	5	.349	121.64	60.	22.34	60.		
123	5	.349	155.91	60.	43.19	60.		
124	5	.349	24.09	120.	43.19	60.		
125	5	.349	55.35	120.	22.34	60.		
126	5	.349	61.25	120.	13.47	60.		
127	5	.349	99.74	120.	13.47	60.		
128	5	.349	121.64	120.	22.34	60.		
129	5	.349	155.91	120.	43.19	60.		
130	5	.349	24.09	180.	43.19	60.		
131	5	.349	52.35	180.	22.34	60.		
132	5	.349	61.25	180.	13.47	60.		
133	5	.349	99.74	180.	13.47	60.		
134	5	.349	121.64	180.	22.34	60.		
135	5	.349	155.91	180.	43.19	60.		
136	5	.349	24.09	240.	43.19	60.		
137	5	.349	55.35	240.	22.34	60.		
138	5	.349	61.25	240.	13.47	60.		
139	5	.349	99.74	240.	13.47	60.		
140	5	.349	121.64	240.	22.34	60.		
141	5	.349	155.91	240.	43.19	60.		
142	5	.349	24.09	300.	43.19	60.		
143	5	.349	55.35	300.	22.34	60.		
144	5	.349	61.25	300.	13.47	60.		
145	5	.349	99.74	300.	13.47	60.		
146	5	.349	121.64	300.	22.34	60.		
147	5	.349	155.91	300.	43.19	60.		
148	TGAST REFLET=1., MT=3., L=1., NR=10, NC=10 S							

C#FD	1	11	21	31	41	51	61	71
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FIGURE 5-1 - Concluded

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TABLE 5-1.  $\rho$ ,  $c_p$ ,  $k$  FOR WATER AND TEFLON

	$\rho$ (1b/ft <sup>3</sup> )	$c_p$ Btu/1b/°R)	$k$ Btu/ft/°R/sec)	T(°R)
Water (1/8-in. Thick)	62.4	0.5	0.323	480.00
	62.4	0.5	0.329	491.49
	62.4	144.5	0.329	491.50
	62.4	145.0	0.329	492.50
	62.4	1.0	0.329	492.51
	62.4	1.0	0.334	500.00
Teflon (1/16-in. Thick Total)	135.0	0.229	$0.414 \times 10^{-4}$	480.00
	135.0	0.234	$0.414 \times 10^{-4}$	491.49
	135.0	0.234	$0.414 \times 10^{-4}$	491.50
	135.0	0.234	$0.414 \times 10^{-4}$	492.50
	135.0	0.234	$0.414 \times 10^{-4}$	492.51
	135.0	0.238	$0.415 \times 10^{-4}$	500.00

## PROGRAM BASIC

TRANSFECTORY DATA WILL BE INPUT VIA CARDS

FROM HEAT WILL CALCULATE THERMAL DATA

SELECT WILL PROVIDE DIRECTED WITH OPTICAL PROPERTIES DATA

ANALYFC WILL CALCULATE TARGET SIGNATURE DATA

NO NATURAL PLUME, OR NUCLEAR BACKGROUNDF DATA WILL BE INPUT OR CALCULATED

NO OFFAXIS CALCULATIONS WILL BE MADE

NO SENSOK MODEL CALCULATIONS WILL BE MADE

TARGET SYNTHESIS PERFORMED

NORMAND = 3

LAMBDA INCREMENT WAS NOT INPUT. DEFAULTED TO .01

LWAVE	191	LAMINC = .10 - 1E+10				
.504000E+01	.510600E+01	.520000E+01	.530000E+01	.540000E+01	.550000E+01	.570000E+01
.580000E+01	.590000E+01	.600000E+01	.610000E+01	.620000E+01	.630000E+01	.650000E+01
.660000E+01	.670000E+01	.680000E+01	.690000E+01	.700000E+01	.710000E+01	.730000E+01
.740000E+01	.750000E+01	.760000E+01	.770000E+01	.780000E+01	.790000E+01	.810000E+01
.820000E+01	.830000E+01	.840000E+01	.850000E+01	.860000E+01	.870000E+01	.890000E+01
.900000E+01	.910000E+01	.920000E+01	.930000E+01	.940000E+01	.950000E+01	.970000E+01
.980000E+01	.990000E+01	.100000E+02	.101000E+02	.102000E+02	.103000E+02	.105000E+02
.106000E+02	.107000E+02	.108000E+02	.109000E+02	.110000E+02	.111000E+02	.113000E+02
.114000E+02	.115000E+02	.116000E+02	.117000E+02	.118000E+02	.119000E+02	.121000E+02
.122000E+02	.123000E+02	.124000E+02	.125000E+02	.126000E+02	.128000E+02	.129000E+02
.131000E+02	.132000E+02	.132000E+02	.133000E+02	.134000E+02	.135000E+02	.137000E+02
.138000E+02	.139000E+02	.140000E+02	.141000E+02	.142000E+02	.143000E+02	.145000E+02
.146000E+02	.147000E+02	.148000E+02	.149000E+02	.150000E+02	.151000E+02	.153000E+02
.154000E+02	.155000E+02	.156000E+02	.157000E+02	.158000E+02	.160000E+02	.161000E+02
.162000E+02	.163000E+02	.164000E+02	.165000E+02	.166000E+02	.167000E+02	.169000E+02
.171000E+02	.172000E+02	.173000E+02	.174000E+02	.175000E+02	.177000E+02	.179000E+02
.183000E+02	.185000E+02	.186000E+02	.187000E+02	.188000E+02	.189000E+02	.190000E+02
.186000E+02	.187000E+02	.188000E+02	.189000E+02	.190000E+02	.191000E+02	.193000E+02
.194000E+02	.195000E+02	.196000E+02	.197000E+02	.198000E+02	.199000E+02	.201000E+02
.202000E+02	.203000E+02	.204000E+02	.205000E+02	.206000E+02	.207000E+02	.209000E+02
.210000E+02	.211000E+02	.212000E+02	.213000E+02	.214000E+02	.215000E+02	.217000E+02
.214000E+02	.215000E+02	.216000E+02	.217000E+02	.218000E+02	.220000E+02	.224000E+02
.226000E+02	.227000E+02	.228000E+02	.229000E+02	.230000E+02	.231000E+02	.233000E+02
.227000E+02	.228000E+02	.229000E+02	.230000E+02	.231000E+02	.232000E+02	.234000E+02

FIGURE 5-2. BALLOON EXAMPLE OUTPUT

MODE	1	(1=INST.FLUX, 2=ROLL-AVGEN FLUX)
MONTH	9	
DAY	5	
GMT	0.00	0.00
SUN-LAT,LNG	0.00	0.00
ATIM. ANGLES	9.00	60.00
POLAR ANGLES	33.56	66.00
EXOCHEAR		
PIECES	36	
ATIM. DIVISIONS	5	
LONG. DIVISIONS	6	

TIME	CONDITION	AVERAGE FLUXES -- W / (M <sup>2</sup> · SR)	
		SUN / ALBEDO	EARTH
6.00000	SUNLT	344.9613	54.42691
12.000	SUNLT	344.9610	54.42691
18.000	SUNLT	344.9610	54.42691
24.000	SUNLT	344.9610	54.42691
30.000	SUNLT	344.9610	54.42691
36.000	SUNLT	344.9610	54.42691
42.000	SUNLT	344.9610	54.42691
48.000	SUNLT	344.9610	54.42691
54.000	SUNLT	344.9610	54.42691
60.000	SUNLT	344.9610	54.42691
66.000	SUNLT	344.9610	54.42691
72.000	SUNLT	344.9610	54.42691
78.000	SUNLT	344.9610	54.42691
84.000	SUNLT	344.9610	54.42691
90.000	SUNLT	344.9610	54.42691
96.000	SUNLT	344.9610	54.42691
102.000	SUNLT	344.9610	54.42691
108.000	SUNLT	344.9610	54.42691
114.000	SUNLT	344.9610	54.42691
120.000	SUNLT	344.9610	54.42691
126.000	SUNLT	344.9610	54.42691
132.000	SUNLT	344.9610	54.42691
138.000	SUNLT	344.9610	54.42691
144.000	SUNLT	344.9610	54.42691
150.000	SUNLT	344.9610	54.42691
156.000	SUNLT	344.9610	54.42691
162.000	SUNLT	344.9610	54.42691
168.000	SUNLT	344.9610	54.42691
174.000	SUNLT	344.9610	54.42691
180.000	SUNLT	344.9610	54.42691

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FROM CORTA ESTABLISHED TO DDC

EIN									
ENUT					ELF				
STAT AREA	PAC	FAUTS	EIN	TUNL TH	CONJ	TR-10	PAO	TRASS	ELF
1	3.76	.553	.033	1.81	.547	1.92	2.05	115F-01	.901
2	3.76	.466	.833	2.4	1.64	3.44	3.05	115F-02	.901
3	3.76	.765	.157	1.57	2.73	4.62	1.05	115F-02	.901
4	3.76	.986	.167	3.23	3.63	5.74	3.05	115F-02	.901
5	3.75	.565	.167	2.51	2.81	5.87	3.05	115F-02	.901
6	3.76	.553	.033	1.61	6.01	6.39	3.05	115F-02	.901
7	3.76	.553	.033	1.81	.547	1.92	3.15	115F-02	.901
8	3.76	.465	.502	2.46	1.64	3.44	3.05	115F-02	.901
9	3.76	.486	.167	1.23	2.73	4.61	3.05	115F-02	.901
10	3.75	.985	.167	3.23	3.63	5.74	3.15	115F-02	.901
11	3.76	.966	-.540	2.84	4.92	6.87	3.05	115F-02	.901
12	3.75	.553	-.033	1.81	6.01	6.39	3.05	115F-02	.901
13	3.76	.553	-.033	1.81	.547	1.92	3.05	115F-02	.901
14	3.75	.466	.502	2.46	1.64	3.44	3.05	115F-02	.901
15	3.76	.595	.167	1.23	2.73	4.61	3.05	115F-02	.901
16	3.76	.986	.167	3.23	3.63	5.74	3.05	115F-02	.901
17	3.76	.866	-.546	2.84	4.92	6.87	3.05	115F-02	.901
18	3.75	.553	-.033	1.81	6.01	6.39	3.05	115F-02	.901
19	3.76	.553	-.033	1.81	.547	1.92	3.05	115F-02	.901
20	3.76	.466	.502	2.46	1.64	3.44	3.05	115F-02	.901
21	3.76	.985	.167	1.23	2.73	4.61	3.05	115F-02	.901
22	3.76	.986	.167	3.23	3.63	5.74	3.05	115F-02	.901
23	3.75	.466	-.546	2.84	4.92	6.87	3.05	115F-02	.901
24	3.76	.553	-.033	1.81	6.01	6.39	3.05	115F-02	.901
25	3.76	.553	-.033	1.81	.547	1.92	3.05	115F-02	.901
26	3.76	.466	.502	2.46	1.64	3.44	3.05	115F-02	.901
27	3.75	.985	.167	1.23	2.73	4.61	3.05	115F-02	.901
28	3.76	.986	.167	3.23	3.63	5.74	3.05	115F-02	.901
29	3.76	.466	-.546	2.84	4.92	6.87	3.05	115F-02	.901
30	3.76	.553	-.033	1.81	6.01	6.39	3.05	115F-02	.901
31	3.75	.553	-.033	1.81	.547	1.92	3.05	115F-02	.901
32	3.76	.466	.502	2.46	1.64	3.44	3.05	115F-02	.901
33	3.75	.985	.167	1.23	2.73	4.61	3.05	115F-02	.901
34	3.76	.986	.167	3.23	3.63	5.74	3.05	115F-02	.901
35	3.76	.466	-.546	2.84	4.92	6.87	3.05	115F-02	.901
36	3.76	.553	-.033	1.81	6.01	6.39	3.05	115F-02	.901

FIGURE 5-2 - Continued

STAT APPENDIX

5-8

**100% FROM THE BEST QUALITY PRACTICABLE  
FROM COPY REVISED TO DDC**

三  
卷之三

506.122  
492.095

FIGURE 5-2 - Continued

**FIGURE 5-2** = Continued

TEMP  
MAX  
MIN

56 J-269  
221.005

FIGURE 5-2 - Continued

三  
卷

566-125  
492.0125

THIS PAGE IS THE QUALITY PRACTICABLE  
FROM COPIER FURNISHED TO DDC

**FIGURE 5-2** = Concluded

## 5.2 REPLICA

This example consists of a biconic shape with a tangentially fitting spherical nosecap that is divided into 24 stations, eight longitudinal by three azimuthal. The nose is made of silicon phenolic that tapers from 0.3 in. for the first two nose stations (along a single longitudinal ray) to 0.2 in. for the third nose station. The first frustum has two stations, each consisting of a 0.1-in. aluminum structure covered by a 0.1-in. thickness of silicon phenolic. The second frustum contains two stations with a 0.1-in.-thick aluminum structure covered by 0.05 in. of silicon phenolic. The base has one station of material structured similar to that of the first frustum. The dimensions of the replica are shown in Figure 5-3. Note that whenever the material identification code word JMAT of card 6.1 changes, card set 7 must be input. The nose is completely silicon phenolic (JMAT = 203), and the numbers JMAT = 180 and JMAT = 190 have been arbitrarily assigned to the aluminum and silicon phenolic combinations. Equations 3-31 through 3-33 were used to compute the average values of  $\rho$ ,  $c_p$ , and  $k$  that are input. Figure 5-4 and 5-5 show the input and output, respectively, for this example.

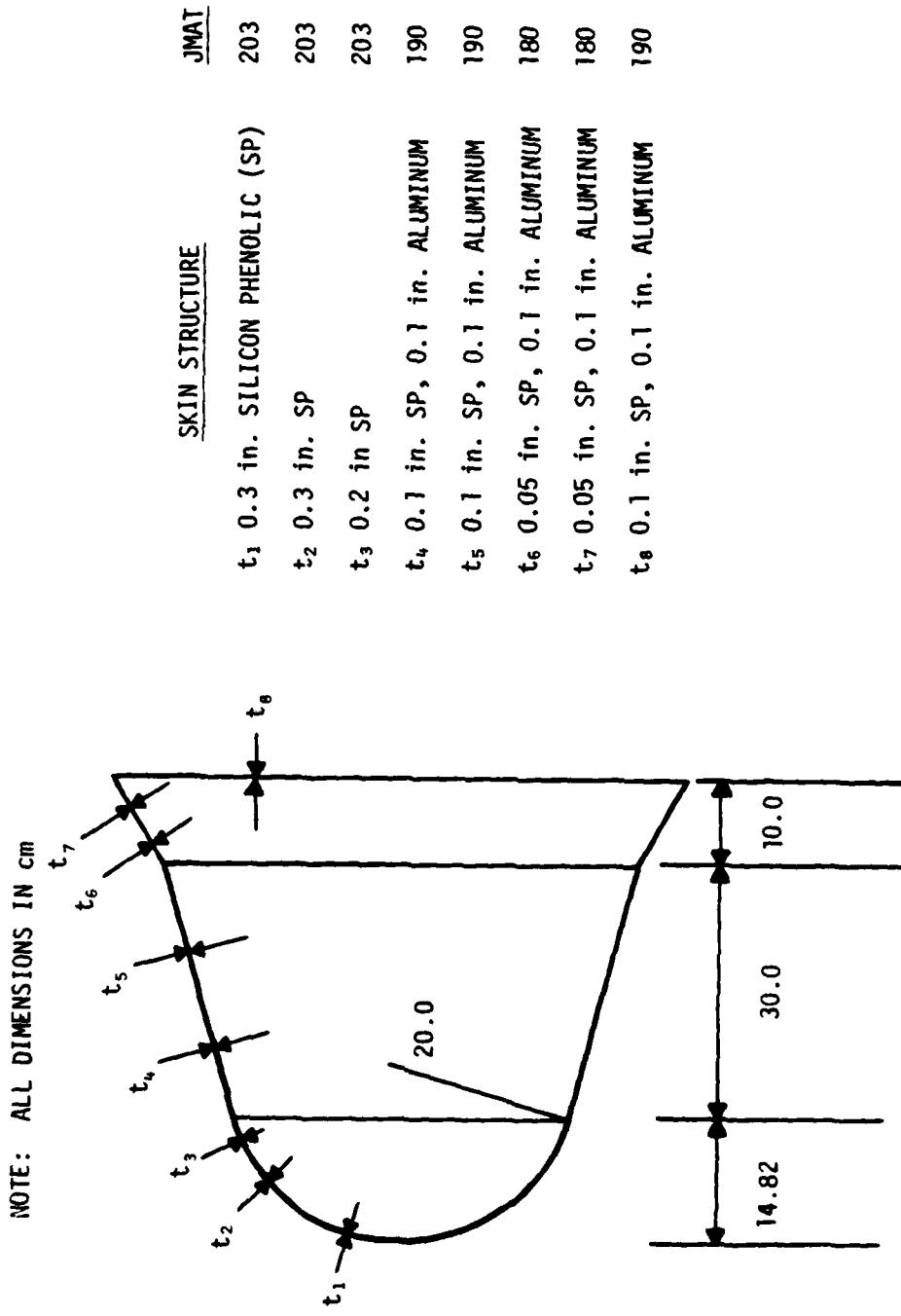


FIGURE 5-3. REPLICA GEOMETRY

```

1 11 21 51 61 71
2 10.4 CALC WARR EXPWAT RATE: CALC SIRWA RAC
3 TACSYN YES
4 3
5 5. 9.
6 11. 15.
7 18. 24.
8
9 SMALL
10 3 2.07 14.42 3
11 2 15.6
12 2 15.7 5.01
13 2 16.7 1.01
14 1.01
15 1.01
16 1.01
17 PV EXA4LL 11.0 2 1.0 15.0
18 FV-USC4 300. 1 161.115 59.814 46.597 216.40 -2.137
19 1. 150. 161.115
20 Ga-neg4
21 54.957 -90.491
22 F4U
23
24 EXIT
25 $SALAR,TYPE=1,SLATE=0,SMCUTTH=0,POWDAV=1,RTCTn=3
26
27 . 120. 24.
28 28.78 50.99 67.51 75.0
29
30
31 $7.99 ALL DALLAS. $
32 F 7
33 233.250 50.0
34 263.250 50.0
35 263.367 50.0
36 197.167 50.0
37 193.367 50.0
38 183.0125 50.0
39 164.0125 50.0
40 194.167 50.0
41 203.0250 50.0
42 233.0250 50.0
43 263.0167 50.0
44 497.0167 50.0
45 191.0167 50.0
46 184.0125 50.0
47 193.0125 50.0
48 194.0167 50.0
49 233.0250 50.0
50 233.0250 50.0

```

FIGURE 5-4. REPLICA EXAMPLE INPUT

Call No.	1	11	21	31	41	51	61	71
51	257	.167	507					
52	190	.167	567					
53	191	.167	567					
54	180	.125	547					
55	180	.125	547					
56	191	.167	567					
57	4	139,r	75					
58	44,r		179					
59	51,r		274					
60	51,r		221					
61	74,r		274					
62	4	151,r,3	75					
63	44,r		193					
64	54,r		213					
65	61,r		216					
66	71,r		226					
67	4	139,r,6	75					
68	44,r		179					
69	51,r		274					
70	61,r		221					
71	71,r		273					
72	4	139,r,6	75					
73	44,r		178					
74	52,r		274					
75	61,r		221					
76	71,r		273					
77	4	150,r,3	75					
78	44,r		194					
79	54,r		225					
80	61,r		216					
81	71,r		226					
82	4	139,r,6	75					
83	44,r		174					
84	52,r		274					
85	61,r		221					
86	71,r		273					
87	4	139,r,6	75					
88	44,r		178					
89	50,r		216					
90	60,r		221					
91	71,r		223					
92	4	150,r,7	75					
93	44,r		193					
94	52,r		224					
95	60,r		215					
96	70,r		216					
97	4	139,r	75					
98	44,r		178					
99	50,r		216					
100	60,r		221					
	1	11	21	31	41	51	61	71

FIGURE 5-4 - Continued

USAGE OF DATA-TRANS...

Car 0	1	11	21	31	41	51	61	71
1-1	*	*	*	*	*	*	*	*
			2121	1121	1121	1121	1121	1121
Car 1	1	11	21	31	41	51	61	71
*	*	*	*	*	*	*	*	*

FIGURE 5-4 - Concluded

PROGRAM BASIC

3BALLS WILL CALCULATE TRAJECTORY DATA  
 EXOHEAT WILL CALCULATE THERMAL DATA  
 SELECT WILL PROVIDE STOREDC MATH OPTICAL PROPERTIES DATA  
 QMIREC WILL CALCULATE TARGET SIGNATURE DATA  
 NO NATURAL, PLUME, OR NUCLEAR BACKGROUND DATA WILL BE INPUT OR CALCULATED  
 NO JEFFAKIS CALCULATIONS WILL BE MADE  
 NO SENSOP MODEL CALCULATIONS WILL BE MADE  
 TARGET SYNTHESIS PERFORMED

WARNING = 3

LAMBDA INCREMENT WAS NOT INPUT. DEFAULTED TO 6.1

LWAVE = 191	LAMINC = *1E4*E+00	LAMINC = *5E4*E+01	LAMINC = *5E5*E+01	LAMINC = *5E6*E+01
*50000E+01	*51000E+01	*52000E+01	*53000E+01	*54000E+01
*54000E+01	*59000UF+01	*60000UF+01	*61000F+01	*62000F+01
*65000CE+01	*67000CE+01	*68000LE+01	*69000CE+01	*70000CE+01
*75000LE+01	*75000LE+01	*76000CE+01	*76000CE+01	*76000CE+01
*82000CE+01	*83000CE+01	*84000CE+01	*85000CE+01	*86000CE+01
*90000CE+01	*92000E+01	*93000E+01	*94000E+01	*95000E+01
*97000E+01	*98000E+01	*10000E+02	*10200E+02	*10400E+02
*10600E+02	*10700E+02	*10800E+02	*10900E+02	*11000E+02
*11600E+02	*11700E+02	*11800E+02	*11900E+02	*12000E+02
*12200E+02	*12300E+02	*12400E+02	*12500E+02	*12600E+02
*13000E+02	*13100E+02	*13200E+02	*13300E+02	*13400E+02
*13800E+02	*13900E+02	*14000E+02	*14100E+02	*14200E+02
*14600E+02	*14700E+02	*14800E+02	*14900E+02	*15000E+02
*15400E+02	*15500E+02	*15600E+02	*15700E+02	*15800E+02
*16200E+02	*16300E+02	*16400E+02	*16500E+02	*16600E+02
*17000E+02	*17100E+02	*17200E+02	*17300E+02	*17400E+02
*17800E+02	*17900E+02	*18000E+02	*18100E+02	*18200E+02
*18600E+02	*18700E+02	*18800E+02	*18900E+02	*19000E+02
*19400F+02	*19500F+02	*19600F+02	*19700F+02	*19800F+02
*20200E+02	*20300E+02	*20400E+02	*20500E+02	*20600E+02
*21600E+02	*21700E+02	*21800E+02	*21900E+02	*22000E+02
*22600E+02	*22700E+02	*22800E+02	*22900E+02	*23000E+02
*23400E+02	*23500E+02	*23600E+02	*23700E+02	*23800E+02

WAVELENGTH FORM

5.4E4 WTCRMS IN Q.7.61 W.CRN.

FIGURE 5-5. REPLICA EXAMPLE OUTPUT

INV EXAMPLE		FLIGHT STARTS AT		J-30 SECONDS AND PRACTICALLY IN		1100 FEET OF		15,000 FEET		22,000 FEET		27,000 FEET	
		000-000	000-000	000-000	000-000	000-000	000-000	000-000	000-000	000-000	000-000	000-000	
RV-0SC4	380.00	1	0	59.00	59.00	84.00	84.00	216.00	216.00	22.00	22.00	-2.16	
RV-0SC4	1500.00	116.00	116.00	59.00	59.00	84.00	84.00	216.00	216.00	22.00	22.00	-2.16	
EV-0SC4	0.00	0	0	59.00	59.00	84.00	84.00	216.00	216.00	22.00	22.00	-2.16	
EV-0SC4	0.00	0	0	59.00	59.00	84.00	84.00	216.00	216.00	22.00	22.00	-2.16	
EVG	0.00	0	0	0	0	0	0	0	0	0	0	0	
EVG	0.00	0	0	0	0	0	0	0	0	0	0	0	
2 Targets Initialized		Alt-MFT		Lat		Long		Vel-FIRS		Ent.Ang Course		Range-KFT	
10 Sec Time		Alt-MFT		Lat		Long		Vel-FIRS		Ent.Ang Course		Range-KFT	
EV-0SC4	0.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	22.000	22.000	357.000	357.000	46.000	
RV-0SC4	0.00	1160.115	59.0004	04.5007	2.00.500	2.00.500	2.00.500	267.000	267.000	41.393	41.393	46.000	
EV-0SC4	30.00	5940.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	212.000	212.000	267.000	267.000	46.000	
RV-0SC4	30.00	1390.099	51.0007	04.0002	2.00.000	2.00.000	2.00.000	252.000	252.000	41.417	41.417	46.000	
EV-0SC4	60.00	5940.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	206.000	206.000	258.000	258.000	46.000	
RV-0SC4	60.00	1626.495	52.0006	04.3007	2.00.300	2.00.300	2.00.300	253.000	253.000	41.400	41.400	46.000	
EV-0SC4	90.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	204.000	204.000	253.000	253.000	46.000	
RV-0SC4	90.00	1845.339	54.0000	04.3007	2.00.300	2.00.300	2.00.300	253.000	253.000	41.400	41.400	46.000	
EV-0SC4	120.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	198.000	198.000	253.000	253.000	46.000	
RV-0SC4	120.00	2050.660	50.0007	04.1001	2.00.100	2.00.100	2.00.100	249.000	249.000	41.400	41.400	46.000	
EV-0SC4	150.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	192.000	192.000	253.000	253.000	46.000	
RV-0SC4	150.00	2250.522	67.0000	04.0003	2.00.000	2.00.000	2.00.000	244.000	244.000	41.400	41.400	46.000	
EV-0SC4	180.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	186.000	186.000	253.000	253.000	46.000	
RV-0SC4	180.00	2446.941	68.0000	04.0003	2.00.000	2.00.000	2.00.000	239.000	239.000	41.400	41.400	46.000	
EV-0SC4	210.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	180.000	180.000	253.000	253.000	46.000	
RV-0SC4	210.00	2625.964	71.0006	03.9012	1.90.000	1.90.000	1.90.000	235.000	235.000	41.400	41.400	46.000	
EV-0SC4	240.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	174.000	174.000	253.000	253.000	46.000	
RV-0SC4	240.00	2797.632	71.3007	03.9021	1.90.000	1.90.000	1.90.000	230.000	230.000	41.400	41.400	46.000	
EV-0SC4	270.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	168.000	168.000	253.000	253.000	46.000	
RV-0SC4	270.00	2959.942	72.0009	03.8003	1.90.000	1.90.000	1.90.000	225.000	225.000	41.400	41.400	46.000	
EV-0SC4	300.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	162.000	162.000	253.000	253.000	46.000	
RV-0SC4	300.00	3113.052	74.0003	03.7005	1.90.000	1.90.000	1.90.000	221.000	221.000	41.400	41.400	46.000	
EV-0SC4	330.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	158.000	158.000	253.000	253.000	46.000	
RV-0SC4	330.00	3256.040	75.0002	03.6007	1.90.000	1.90.000	1.90.000	216.000	216.000	41.400	41.400	46.000	
EV-0SC4	360.00	5900.910	54.0857	-0.00.601	0.0007	0.0010	0.0007	154.000	154.000	253.000	253.000	46.000	
RV-0SC4	360.00	3600.910	76.0001	03.5007	1.90.000	1.90.000	1.90.000	216.000	216.000	41.400	41.400	46.000	

FIGURE 5-5 - Continued

FIGURE 5-5 - Continued

RV-0SC4	368.00	3391.566	760.636	63.316	14780.424	13.414	3580.97	211520.623	-38.673	359.445	31.33	87.74	54.76	
G3-0SC4	390.00	5016.916	54.057	-98.481	295290.475	12.011	350.011	236720.216	-29.842	359.511	24.97	87.95	51.11	
RV-0SC4	392.00	3516.946	77.915	83.272	18241.746	0.510	350.247	236720.216	-29.842	359.511	24.97	87.95	51.11	
G3-0SC4	421.00	5080.910	54.057	-98.481	18241.746	11.691	350.500	236720.216	-29.842	359.511	24.97	87.95	51.11	
RV-0SC4	429.00	1613.249	79.142	83.157	18241.746	11.691	350.414	236720.216	-29.842	359.511	24.97	87.95	51.11	
G3-0SC4	458.00	5980.910	54.057	-98.481	19244.460	10.819	350.585	13703.965	-28.348	359.628	27.10	88.33	62.88	
RV-0SC4	459.00	3746.440	36.437	83.305	19244.460	10.819	350.712	13715.499	-27.676	359.674	27.10	88.52	64.31	
G3-0SC4	470.00	5080.910	54.057	-98.481	19119.572	0.011	350.011	13715.499	-27.676	359.674	27.10	88.52	64.31	
RV-0SC4	470.00	2836.547	41.682	82.851	19119.572	0.011	350.011	13715.499	-27.676	359.674	27.10	88.52	64.31	
G3-0SC4	510.00	5080.910	54.057	-98.481	18241.746	0.010	350.011	14726.656	-26.083	350.772	24.24	90.77	65.76	
RV-0SC4	510.00	1927.504	92.917	83.533	18241.746	0.010	350.011	14726.656	-26.083	350.772	24.24	90.77	65.76	
G3-0SC4	540.00	5980.910	54.057	-98.481	17044.767	6.010	350.011	19236.973	-26.317	150.761	22.78	88.94	67.20	
RV-0SC4	540.00	4007.666	94.164	83.361	17044.767	6.010	350.011	19236.973	-26.317	150.761	22.78	88.94	67.20	
G3-0SC4	574.00	5080.910	54.057	-98.481	17814.667	6.010	350.011	17734.130	-25.674	359.792	21.26	89.16	69.75	
RV-0SC4	574.00	4278.605	52.364	83.167	17814.667	6.010	350.011	17734.130	-25.674	359.792	21.26	89.16	69.75	
G3-0SC4	601.00	5080.910	54.057	-98.481	17736.092	0.011	350.011	17234.712	-25.667	359.817	19.71	90.74	70.30	
RV-0SC4	600.00	6140.618	46.577	83.474	17736.092	0.011	350.011	17234.712	-25.667	359.817	19.71	90.74	70.30	
G3-0SC4	630.00	5980.910	54.057	-98.481	17658.963	0.000	350.011	15732.383	-24.486	359.834	18.13	89.62	71.86	
RV-0SC4	630.00	4193.632	97.786	84.127	17658.963	0.000	350.011	15732.383	-24.486	359.834	18.13	89.62	71.86	
G3-0SC4	660.00	5980.910	54.057	-98.481	17613.254	4.014	350.011	15227.262	-23.938	359.843	16.51	89.96	75.49	
RV-0SC4	660.00	4237.694	94.945	86.755	17613.254	4.014	350.011	15227.262	-23.938	359.843	16.51	89.96	75.49	
G3-0SC4	698.00	5980.910	54.057	-98.481	17568.937	6.016	350.011	15427.443	-22.956	359.855	13.17	90.36	76.93	
RV-0SC4	698.00	4227.646	99.791	236.172	17568.937	6.016	350.011	15427.443	-22.956	359.855	13.17	90.36	76.93	
G3-0SC4	720.00	3980.910	54.057	-98.481	17535.991	6.010	350.011	176.443	-22.956	359.855	13.17	90.36	76.93	
RV-0SC4	720.00	4298.971	98.612	257.774	17535.991	6.010	350.011	176.443	-22.956	359.855	13.17	90.36	76.93	
G3-0SC4	753.00	5980.910	54.057	-98.481	17514.402	1.012	350.011	177.281	-22.525	359.916	11.45	91.63	78.57	
RV-0SC4	753.00	4316.214	37.419	259.721	17514.402	1.012	350.011	177.281	-22.525	359.916	11.45	91.63	78.57	
G3-0SC4	780.00	5980.910	54.057	-98.481	1750.415	0.011	350.011	179.735	-22.146	359.786	9.619	95.91	90.35	
RV-0SC4	780.00	4324.506	45.227	259.825	1750.415	0.011	350.011	179.735	-22.146	359.786	9.619	95.91	90.35	
G3-0SC4	811.00	5980.910	54.057	-98.481	1750.525	-0.555	350.011	179.686	13650.413	-21.916	359.743	7.05	91.19	A2.19
RV-0SC4	811.00	4323.606	95.034	260.029	1750.525	-0.555	350.011	179.686	13650.413	-21.916	359.743	7.05	91.19	A2.19
G3-0SC4	840.00	5980.910	54.057	-98.481	17517.655	-1.529	350.011	179.416	13110.1.9	-21.549	359.697	5.9	91.49	A4.9
RV-0SC4	840.00	4316.325	93.541	260.13	17517.655	-1.529	350.011	179.416	13110.1.9	-21.549	359.697	5.9	91.49	A4.9

FIGURE 5-5 - Continued

## EXPHAT

24 DEGREES  
3 ATM. DIVISIONS  
8 LONG. DIVISIONS

## A74. ANGLES

	0.00	126.00	246.00						
POLAR ANGLES	24.74	58.99	67.51	73.00	76.00	67.00	60.00	19.00	

MODE 1 (1=TNST,FLUX, 2=ROLL-AVRED FLUX)

MONTH 8

DAY 1<sup>st</sup>

GMT 0.00

SUN-LAT,LNG 6.00 0.00

TIME	CONDITION	AVERAGE FLUXES -- W/(M <sup>2</sup> )		MOLECULAR
		SUNVALREDO	EARTH	
30.0000E	SUNLIT	348.6558	128.7528	0.
60.0000E	SUNLIT	348.5368	126.6460	0.
90.0000E	SUNLIT	348.4320	114.1127	0.
120.0000E	SUNLIT	348.3318	108.1425	0.
150.0000E	SUNLIT	348.2341	102.9933	0.
180.0000E	SUNLIT	348.1297	98.17172	0.
210.0000E	SUNLIT	348.1234	93.69156	0.
240.0000E	SUNLIT	347.9122	89.46573	0.
270.0000E	SUNLIT	347.7941	85.84116	0.
300.0000E	SUNLIT	347.6683	81.56494	0.
330.0000E	SUNLIT	347.5425	78.5927	0.
360.0000E	SUNLIT	347.4137	78.24114	0.
390.0000E	SUNLIT	347.2815	76.16533	0.
420.0000E	SUNLIT	347.1512	74.53495	0.
450.0000E	SUNLIT	347.1161	72.93991	0.
480.0000E	SUNLIT	346.8767	71.10149	0.
510.0000E	SUNLIT	346.7342	69.97556	0.
540.0000E	SUNLIT	346.5976	69.09332	0.
570.0000E	SUNLIT	346.4458	67.93349	0.
600.0000E	SUNLIT	346.2975	67.18211	0.
630.0000E	SUNLIT	346.1443	66.66744	0.
660.0000E	SUNLIT	346.1013	66.16748	0.
690.0000E	SUNLIT	346.8664	65.78464	0.
720.0000E	SUNLIT	345.7052	64.71912	0.
750.0000E	SUNLIT	345.3639	65.26899	0.
780.0000E	SUNLIT	345.2271	65.11285	0.
810.0000E	SUNLIT	345.1555	64.9722	0.
840.0000E	SUNLIT	344.8723	64.75417	0.
870.0000E	SUNLIT	344.6619	65.7251	0.
900.0000E	SUNLIT	344.4688	65.31422	0.
930.0000E	SUNLIT	344.2884	65.58016	0.
960.0000E	SUNLIT	344.0878	66.16755	0.
990.0000E	SUNLIT	343.8754	66.68854	0.
1020.0000E	SUNLIT	343.6627	67.59173	0.
1050.0000E	SUNLIT	343.4481	68.42413	0.
1080.0000E	SUNLIT	343.2294	69.33975	0.
1110.0000E	SUNLIT	343.1142	71.37516	0.
1140.0000E	SUNLIT	342.7743	71.85412	0.
1170.0000E	SUNLIT	342.5274	73.15459	0.
1200.0000E	SUNLIT	342.2778	74.72723	0.
1230.0000E	SUNLIT	342.0177	76.39432	0.
1260.0000E	SUNLIT	341.7548	74.76244	0.
1290.0000E	SUNLIT	341.4791	71.32214	0.
1320.0000E	SUNLIT	341.1988	70.11873	0.
1350.0000E	SUNLIT	340.9063	66.14510	0.
1380.0000E	SUNLIT	340.6124	69.66132	0.
1410.0000E	SUNLIT	340.3191	72.07471	0.
1440.0000E	SUNLIT	340.0174	76.24317	0.
1470.0000E	SUNLIT	339.7077	79.59559	0.
		339.4169	1.2.2723	0.
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FIGURE 5-5 - Continued

5-21

QUALITY PRACTICABLE  
EQUIPMENT AND MATERIALS

STAT	APEA	PERP	RADIUS	7ED	RUNLTH	TRASS	?AD	?CND	EIN	EOUT
1	.223	.876	.716	.81E-01	.33	.11E	.981E-04	.58E-05	.125E+04	.55J
2	.223	.629	.51F	.263	.58E	.120	.981E-04	.339E-04	.119E+04	.75G
3	.223	.924	.392	.405	.60F	.773	.667E-01	.131E-04	.78E-	.75G
4	.747	.966	.259	.732	.70F	.11	.981E-04	.208E-03	.44E-	.75G
5	.887	.966	.259	.812	.1.22	.62	.24AE-03	.423E-03	.44E-	.260
6	.375	.466	.59J	.645	.1.55	.47	.135E-03	.149E-02	.252	.260
7	.142	.666	.5JL	.772	.1.72	.16	.116E-03	.274E-02	.51.9	.260
8	1.24	-30.8E-09	-1.3J	.547	1.86	2.8	.116E-03	.231E-02	.67.6	.200
9	.223	.481	.876	.316	.81F-01	.330	.981E-04	.580E-05	.125E+04	.56G
10	.223	.777	.629	.51U	.263	.804	.110	.981E-04	.199E-04	.119E+04
11	.223	.924	.392	.362	.605	.773	.46E-01	.981E-04	.331E-04	.75G
12	.747	.966	.259	.70C	.70C	.11	.367	.230E-03	.623E-03	.75G
13	.447	.966	.259	.P12	1.22	.62	.437	.24AE-03	.14AE-02	.260
14	.375	.866	.51F	.945	1.55	.97	.147	.106E-03	.274E-02	.75G
15	.412	.866	.51L	.1.64	1.72	2.16	.152	.116E-03	.231E-02	.75G
16	1.24	-30.8E-09	-1.06	.543	1.80	2.80	.6.9	.680E-03	.590E-	.200
17	.223	.641	.876	.316	.81E-01	.330	.136	.981E-04	.580E-05	.125E+04
18	.223	.777	.629	.51L	.263	.584	.137	.981E-04	.199E-04	.75G
19	.223	.924	.392	.666	.465	.773	.467E-01	.981E-04	.131E-04	.75G
20	.747	.866	.259	.70C	.732	1.11	.367	.228E-03	.423E-03	.56G
21	.487	.966	.259	.832	1.22	.62	.437	.248E-03	.149E-02	.252
22	.375	.865	.50L	.945	1.55	1.97	.147	.066E-03	.274E-02	.51.9
23	.412	.966	.50C	1.04	1.72	2.16	.162	.116E-03	.231E-02	.75G
24	1.24	-30.8E-09	-1.30	.543	1.84	2.80	.669	.352E-03	.690E-03	.56G

#### TIME DEVELOPMENT OF TEMPERATURES

LONGITUDINAL CONDUCTION TREATED --

OPEN HEAD SURFACE --

MR MTX INV--

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FIGURE 5-5 - Continued

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21  
TEMES  
MAX 571.272  
MIN 472.653

STATE	11	12	13	14	15	16	17	18	19	20
Ala.	549,419	546,141	546,011	546,011	546,011	546,011	546,011	546,011	546,011	546,011
Ala., cont.	528,446	528,733	528,733	528,733	528,733	528,733	528,733	528,733	528,733	528,733
Ala., cont.	526,465	527,582	527,582	527,582	527,582	527,582	527,582	527,582	527,582	527,582
Ala., cont.	522,446	526,367	526,245	526,245	526,245	526,245	526,245	526,245	526,245	526,245
Ala., cont.	520,465	523,076	523,151	523,151	523,151	523,151	523,151	523,151	523,151	523,151
Ala., cont.	518,465	522,923	522,756	522,756	522,756	522,756	522,756	522,756	522,756	522,756
Ala., cont.	516,465	523,032	523,727	523,727	523,727	523,727	523,727	523,727	523,727	523,727
Ala., cont.	514,465	523,452	524,452	524,452	524,452	524,452	524,452	524,452	524,452	524,452
Ala., cont.	512,465	524,457	524,197	524,197	524,197	524,197	524,197	524,197	524,197	524,197
Ala., cont.	510,465	524,274	524,017	524,017	524,017	524,017	524,017	524,017	524,017	524,017
Ala., cont.	508,465	524,542	525,455	525,455	525,455	525,455	525,455	525,455	525,455	525,455
Ala., cont.	506,465	525,127	527,855	527,855	527,855	527,855	527,855	527,855	527,855	527,855
Ala., cont.	504,465	523,723	522,743	522,187	522,187	522,187	522,187	522,187	522,187	522,187
Ala., cont.	502,465	527,346	526,545	526,297	526,297	526,297	526,297	526,297	526,297	526,297
Ala., cont.	500,465	526,497	526,397	526,777	526,777	526,777	526,777	526,777	526,777	526,777
Ala., cont.	498,465	526,263	522,633	522,459	522,459	522,459	522,459	522,459	522,459	522,459
Ala., cont.	496,465	519,476	522,147	521,466	521,466	521,466	521,466	521,466	521,466	521,466
Ala., cont.	494,465	517,395	521,416	521,416	521,416	521,416	521,416	521,416	521,416	521,416
Ala., cont.	492,465	515,843	519,353	519,199	519,199	519,199	519,199	519,199	519,199	519,199
Ala., cont.	490,465	514,426	518,602	518,703	518,703	518,703	518,703	518,703	518,703	518,703
Ala., cont.	488,465	513,429	517,464	517,137	517,137	517,137	517,137	517,137	517,137	517,137
Ala., cont.	486,465	512,259	516,613	516,961	516,961	516,961	516,961	516,961	516,961	516,961
Ala., cont.	484,465	511,115	515,749	514,920	514,920	514,920	514,920	514,920	514,920	514,920
Ala., cont.	482,465	509,904	514,524	512,917	512,917	512,917	512,917	512,917	512,917	512,917
Ala., cont.	480,465	510,495	513,463	512,425	512,425	512,425	512,425	512,425	512,425	512,425
Ala., cont.	478,465	507,437	512,666	511,954	511,235	511,235	511,235	511,235	511,235	511,235
Ala., cont.	476,465	506,790	511,324	511,316	511,316	511,316	511,316	511,316	511,316	511,316
Ala., cont.	474,465	505,764	511,313	510,379	510,466	510,466	510,466	510,466	510,466	510,466
Ala., cont.	472,465	504,703	504,417	509,171	509,171	509,171	509,171	509,171	509,171	509,171
Ala., cont.	470,465	503,424	509,224	508,224	508,224	508,224	508,224	508,224	508,224	508,224
Ala., cont.	468,465	502,484	508,362	507,461	507,461	507,461	507,461	507,461	507,461	507,461
Ala., cont.	466,465	501,465	507,443	504,420	506,462	506,462	506,462	506,462	506,462	506,462
Ala., cont.	464,465	501,834	506,759	505,759	505,759	505,759	505,759	505,759	505,759	505,759
Ala., cont.	462,465	500,465	505,464	504,463	505,116	505,116	505,116	505,116	505,116	505,116
Ala., cont.	460,465	499,598	515,114	508,147	503,431	504,465	504,465	504,465	504,465	504,465
Ala., cont.	458,465	498,529	516,512	516,421	512,233	513,376	513,376	513,376	513,376	513,376
Ala., cont.	456,465	497,613	513,626	522,741	524,532	524,532	524,532	524,532	524,532	524,532
Ala., cont.	454,465	497,446	512,923	511,904	511,773	512,441	512,441	512,441	512,441	512,441
Ala., cont.	452,465	496,336	512,214	505,711	511,127	511,127	511,127	511,127	511,127	511,127
Ala., cont.	450,465	495,461	515,151	504,457	512,426	511,274	511,274	511,274	511,274	511,274
Ala., cont.	448,465	494,373	506,727	501,119	501,119	501,119	501,119	501,119	501,119	501,119
Ala., cont.	446,465	493,339	501,287	499,473	511,081	511,081	511,081	511,081	511,081	511,081
Ala., cont.	444,465	492,715	498,694	498,426	508,037	509,501	509,501	509,501	509,501	509,501
Ala., cont.	442,465	491,166	498,124	496,273	498,421	499,351	499,351	499,351	499,351	499,351
Ala., cont.	440,465	492,438	498,594	497,740	497,474	498,566	513,374	498,566	513,374	513,374
Ala., cont.	438,465	491,135	498,048	497,257	497,481	498,319	502,623	502,623	502,623	502,623
Ala., cont.	436,465	491,567	497,063	496,742	497,146	497,648	502,234	502,234	502,234	502,234
Ala., cont.	434,465	491,231	497,144	496,742	496,636	497,252	501,141	502,547	501,540	514,078
Ala., cont.	432,465	490,837	496,724	495,927	492,243	496,480	501,484	521,681	511,687	575,177
Ala., cont.	430,465	490,477	496,323	495,549	495,493	496,493	520,225	520,427	511,411	577,171
Ala., cont.	428,465	492,154	495,986	495,192	495,671	491,309	496,707	520,417	520,407	577,171

FIGURE 5-5 - Continued

21  
TELEGRAMS  
MAY 571-271  
MIN 472-653

NAME/STAT	21	22	23	24
6.***	503.036	544.150	546.221	546.416
33.***	501.345	504.630	501.571	519.557
586.***	500.456	503.345	502.855	513.152
46.***	500.707	504.500	503.057	517.205
126.***	505.703	506.500	506.401	516.101
150.***	507.822	506.500	505.620	515.500
150.***	504.150	507.421	506.270	514.537
180.***	500.835	508.489	506.815	514.113
240.***	507.222	508.857	507.216	513.747
270.***	501.119	509.422	507.678	512.684
300.***	501.988	504.957	507.950	512.916
310.***	502.776	500.466	498.264	531.150
360.***	493.893	501.791	506.411	517.715
360.***	504.175	501.144	504.577	531.198
420.***	505.797	505.445	508.708	529.516
420.***	505.369	501.953	506.816	524.931
480.***	505.495	501.929	508.876	528.776
510.***	506.379	502.123	504.923	527.941
540.***	506.226	502.296	506.951	527.324
570.***	507.239	502.433	506.942	526.826
630.***	507.416	502.555	506.959	526.342
630.***	497.865	502.689	506.948	495.477
680.***	504.247	501.764	504.921	535.429
690.***	505.495	502.021	508.749	529.496
720.***	500.851	502.890	504.953	524.573
750.***	500.126	502.964	506.810	524.164
780.***	509.362	502.994	509.772	523.768
810.***	509.434	503.046	508.723	523.374
860.***	509.862	503.485	506.685	523.015
870.***	506.597	503.327	506.646	522.646
900.***	506.232	502.167	506.466	522.246
930.***	506.391	503.210	506.576	521.953
940.***	506.575	505.255	506.560	521.622
950.***	506.757	503.324	506.537	521.111
1020.***	506.939	505.334	506.519	520.666
1050.***	501.120	506.334	506.514	521.486
1080.***	501.120	503.395	506.525	521.370
1110.***	501.492	502.584	506.525	521.746
1140.***	501.667	503.679	506.577	520.794
1150.***	501.667	503.763	506.622	519.517
1180.***	502.138	503.893	506.586	519.226
1190.***	502.138	504.877	506.586	519.194
1200.***	502.138	504.877	506.756	519.064
1260.***	502.503	505.169	506.864	518.666
1290.***	502.816	505.365	506.964	518.206
1320.***	503.047	506.551	506.975	518.123
1350.***	503.376	506.766	506.945	517.555
1360.***	503.679	506.992	506.933	517.586
1410.***	501.299	505.226	506.506	517.322
1440.***	500.320	505.588	506.708	517.162
1470.***	500.320	505.795	506.716	516.842

EMP E XONCAT

FIGURE 5-5 - Concluded

## 6. HINTS AND DIAGNOSTICS

1. To call MBALL, it is necessary that: HEATRV = EXOHEAT  
TARGSYN = YES  
BALL = 2 or 3 (in NAMELIST/  
IBALL)
2. If the removal of a piece (creation of a hole) is desired:
  - a. The initial temperature of the piece is input as 0.
  - b. The external flux entering the interior is neglected, so the number of stations removed should be small.
3. For a skin that consists of layers of different materials, the appropriate averages of the material properties should be input (see Equations 3-31 through 3-33) in card set 7 (Table 4-3).
4. If, for any reason, it is necessary to model a station with an outside emissivity equal to zero (this piece thus has no radiative communication with the external environment) the outside emissivity on card 7.1 (Table 4-3) should be set to a small number, but not zero. This is because the external emissivity is set equal to zero internally for an open station (when TOLD = 0) and is used as the flag for that open station in the calculations, so that any outside emissivity equal to zero would be interpreted as an open station.

## 7. REFERENCES

1. E. K. Stewart, "Optical Signatures Code, Volume I - Basic Option", Sixth Distribution, Teledyne Brown Engineering, March 1979
2. H. Rose, D. Anding, R. R. Kauth, and J. Walker, "Handbook of Albedo and Earthshine", Environmental Research Institute of Michigan, University of Michigan, Ann Arbor, Michigan, June 1973, 190201-1-T
3. J. V. Beaupre, "Optical Signatures Code, Volume II - Exoatmospheric Thermal Response Model: EXOHEAT", Fifth Distribution, Teledyne Brown Engineering, December 1977